

## Round Robin Test for composite-to-brick shear bond characterization

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**Abstract** The paper presents the experience of a working group within the RILEM Technical Committee 223-MSc ‘Masonry Strengthening with Composite materials’, aimed at developing a standardized, reliable procedure for characterizing the bonding mechanism of masonry elements strengthened with

composite materials under shear actions. Twelve laboratories from European universities and research centers were involved. Two different set-ups were compared, for single-lap and double-lap shear tests (the latter in two versions). Four kinds of fiber fabrics, i.e., glass, carbon, basalt and steel, were applied with

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epoxy resins (wet lay-up system) to clay brick units, for a total of 280 monotonic tests. The results provided information regarding the response of externally bonded-to-brick composites in terms of observed failure mechanisms, load capacity, effective transfer length, and bond shear stress–slip behavior. The test results of the 12 laboratories constitute a set of statistically representative data which may conveniently be used for setting appropriate design provisions and guidelines.

**Keywords** Masonry · Bond · Clay brick · FRP · SRP · Shear test

## 1 Introduction

Composite materials are increasingly proposed for strengthening existing constructions, even in the field of masonry buildings belonging to cultural heritage. In particular, externally bonded (EB) fiber-reinforced systems, for both fiber-reinforced polymer (FRP) and steel-reinforced polymer (SRP) are mostly adopted to strengthen structural components such as walls (both in- and out-of-plane) [10, 11, 25, 29, 34, 35, 41, 44, 47–50, 53, 57, 67–69, 71, 72, 78, 79], to improve performance in arches and vaults (to repair cracks and increase global ductility) [14–16, 18, 19, 27, 32, 33, 39, 40, 52, 54, 60, 77] or to confine columns or pillars [8, 9, 30, 56, 59]. In all these kinds of applications, the advantages of using composites are well-known and include: tensile strength with negligible addition of loads, feasibility and versatility in applications, and corrosion resistance. In addition, the more recent use of steel products, besides fiber-reinforced ones, adds to the above-mentioned advantages the possibility of folding the strips, as well as application with inorganic materials such as matrix (e.g., mortars based on hydraulic binders), to improve compatibility and removability [17, 22, 43, 63–65]. The prospects of other recent mineral fibers (e.g., basalt) or natural ones as reinforcing materials (e.g., flax, hemp) are also promising, to reduce obtrusiveness and improve sustainability [43, 80, 82].

The most critical phenomenon influencing the effectiveness of intervention is debonding of the reinforcing system from its substrate. This is a brittle phenomenon and should therefore be avoided. For this

reason, clarification and characterization of behavior at the composite–masonry interface is essential, and involves two important issues: (i) definition of proper experimental procedures and (ii) identification of suitable parameters to be used in design formulations and assessment. As regards actions perpendicular to the surface, the simple test method proposed by ASTM C1583 [12] to measure pull-off strength is easy to perform both in the laboratory and in situ for quality control, and also provides the reference strength to be used in simplified models for design [77]. Nevertheless, it is the behavior under action parallel to the surface of the substrate which is involved in most common applications on structural components. In this context, despite the very widespread use of composites in construction and structural upgrading, specific design rules are still far from generally agreed upon, as are experimental procedures for parameter characterization. Therefore, although various kinds of research are provided in the literature on this subject, there is a great need for harmonization of test methods, for good reproducibility and reliable comparison of results. This is particularly urgent for masonry structures, especially in the case of applications in the historical field, in which lack of knowledge may severely compromise their preservation [76], and the considerable variability of types and mechanical properties should be taken into account.

As regards codes, the two guidelines available at international level, CNR DT200 [28], released by the Italian Research Council, and ACI 440.7R-10 [3], issued by the American Concrete Institute, are based on different approaches concerning bonding on masonry, adopting for this material the results of studies on reinforced concrete. They propose design parameters evaluated through empirical coefficients related to various factors (fracture energy or ultimate strains, respectively). However, the ACI guidelines do not take into account substrate properties, whereas the CNR ones express the reference factor through masonry strength. These coefficients thus still need proper refinement, from further and specifically oriented experimental campaigns.

As regards testing procedures, many methods have been developed for concrete elements, examples being the single-lap shear test (SLST) [26, 70], double-lap (DL) pull–pull shear test [51, 58], DL push–pull shear test [21] and beam-type test [31]. Regarding masonry, a fundamental contribution toward clarifying these aspects was made by several research groups who



tested bonding on clay bricks [20, 23, 42, 45, 46], stone elements [5, 7, 37] and masonry prisms [24, 61]. The DL push–pull shear test, also known as the double-shear push or near-end supported double-shear test [81], is the most frequently adopted, mainly because it allows a universal testing machine to be used. Nevertheless, its reliability in comparison with the SLST is questionable, due to the difficulty of reproducing specimen symmetry (usually a single brick between two strips of fabric or laminate glued on both sides) and consequently ensuring equal distribution of load between the strips [55]. A basic problem for DL test set-ups concerns the correct alignment of the load on the specimen. This discussion on masonry is still open, and good synergy from researchers is therefore needed, for proper progress in its understanding.

In this connection, the RILEM Technical Committee 223-MS-C ‘Masonry Strengthening with Composite materials’ (TC 223-MS-C) has been working since 2007, aiming at: (i) systematization of current knowledge on the structural behavior of masonry strengthened with composites, including experimental, analytical/modeling works, and collection of case studies; (ii) specification of limitations and capabilities of the various reinforcing systems in different contexts (modern or historical); (iii) identification of the most critical aspects influencing intervention effectiveness and their experimental characterization with reliable procedures; and (iv) proposals for recommendations or guidelines as contributions to pre-standards to clarify specific problems of composites applied to masonry. The TC has more than 45 people belonging to 27 institutions representative of 13 countries. A data warehouse allowing the storage and comparison in real time of data published in the literature has been produced [49], and a comprehensive state-of-the-art report will be finalized. Moreover, as bonding emerged as the most critical problem affecting interventions, a round robin test (RRT) was proposed, focusing on the application of fabrics as EB-FRP/SRP to masonry and behavior under shear actions. In its first phase, the possible influence of mortar bed joints was neglected, so that composites were only applied to units. This was done in order to reduce the number of variables and to keep as the main objective clarification of the influence of some important aspects related to bonding under shear actions, i.e., reinforcing materials, test set-up, bond length, measurement patterns, etc. Twelve institutions

were involved: Cracow University of Technology (Poland), University of Minho (Portugal), University of Patras (Greece), eight Italian university laboratories (University of Cassino and Southern Lazio, University of Chieti-Pescara, University of Naples ‘Federico II’, University of Perugia, University Roma Tre, University of Roma Tor Vergata, University of Salento, and University of Padova), and the applied research center Tecnalia (Spain). Starting from December 2009 for specimen manufacture, tests were concluded by the 12 laboratories in about 6 months.

In this paper, preliminary choices, characterization of basic materials, preparation of specimens, and test execution phases are described. The main experimental results are then discussed, in terms of comparison of performance among various composites and the influence of different aspects.

The final aim of this investigation is the development of a standardized, reliable procedure to study the debonding mechanism of masonry elements strengthened by composite materials and to identify significant parameters for harmonizing laboratory experimental procedures, to be drafted in specific recommendations.

## 2 Experimental work

Eight laboratories from Italy and four from other European countries (Poland, Portugal, Spain, Greece) were involved in the RRT. Each laboratory carried out SLSTs and/or double-lap shear tests (DLST) on specimens reinforced with four types of EB composites, applied with epoxy resins: glass (GFRP), basalt (BFRP), carbon (CFRP) and steel (SRP). A soft mud clay brick was considered as reference for the masonry substrate. Some fixed parameters and conditions were also preliminarily agreed upon: the width of the composite (50 mm), its bonded length (160 mm), measurement patterns (with strain gauges and transducers), and displacement and acquisition rates (preferably 0.005 mm/s and at least 5 Hz, respectively). Load was applied monotonically and measured either by the load cell of the universal machine and/or by additional load cells, with ultimate capacities varying from 20 to 500 kN. Tests were performed in displacement control mode until complete detachment (or rupture) of the composite strip.

SLST were performed by seven institutions, which made their own set-ups; the specimen was composed of a single brick with a composite strip glued to one side. For DLST, the specimen was composed of a single strip reversed into a U-shape and glued at its ends to the two faces of the substrate. Two configurations were tested, in order to assess the influence of the curvature of the reinforcement (since too high curvatures may induce premature rupture of the strip): (i) specimens DL55, composed of a single brick, in which the diameter of the curved part was equal to unit thickness, i.e., 55 mm and (ii) specimens DL110, obtained by gluing together two bricks with a thin layer of resin, giving a double thickness of 110 mm.

DLST were performed by seven institutions (three using the thinner specimen type and four the thicker one), with a set-up especially designed after discussion among members and adapted to the specific conditions in the various laboratories.

Five specimens of each composite, with single and/or DL set-ups, were tested by each laboratory, for

a total of 280 tests, as shown in Table 1. All basic materials (bricks and reinforcing systems) were mechanically characterized in three laboratories during the first phases of the RRT.

## 2.1 Materials

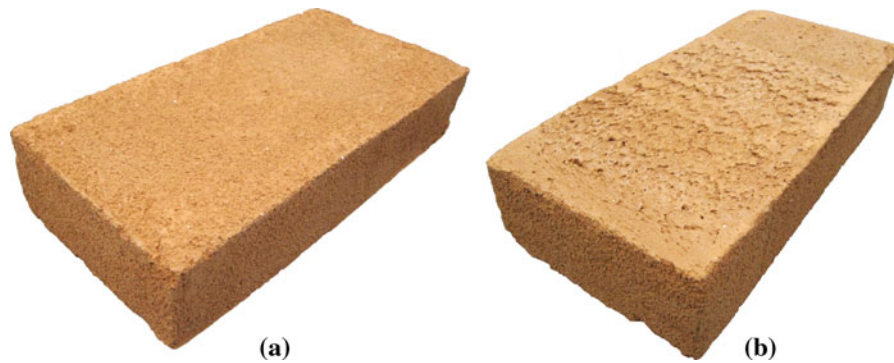
### 2.1.1 Brick properties

Solid facing clay bricks provided from SanMarco-Terreal Italia (Noale, Italy), called “Rosso Vivo—A6R55W”, were used as substrates for all shear tests. They are soft mud bricks (also known as pressed bricks) 250 mm long, 120 mm wide and 55 mm thick, with two surfaces: the top surface, labeled for experiments as “front”, is smoother and more refined than the more porous bottom surface, labeled “back” (Fig. 1).

Compressive and tensile strength, as well as elastic modulus, were characterized at UNIPD and UMINHO, for a total of 36 tests. In detail, three-point

**Table 1** Round robin experimental test matrix

Institutions		Tests performed											
		SLSTs				DLSTs							
Name	Acronym					DL55				DL110			
		GFRP	BFRP	CFRP	SRP	GFRP	BFRP	CFRP	SRP	GFRP	BFRP	CFRP	SRP
Tecnia R&I	TECN	5	5	5	5								
University of Minho	UMINHO	5	5	5	5								
University of Salento	UNILE	5	5	5	5								
University Roma Tre	UNIRM3	5	5	5	5								
University of Patras	UPATRAS	5	5	5	5								
Cracow University of Technology	CUT	5	5	5	5	5	5	5	5				
University of Padova	UNIPD	5	5	5	5	5	5	5	5				
University of Perugia	UNIPG					5	5	5	5				
University of Cassino and Southern Lazio	UNICAS									5	5	5	5
University of Chieti-Pescara	UNICH									5	5	5	5
University of Naples ‘Federico II’	UNINA									5	5	5	5
University of Roma Tor Vergata	UNIRM2									5	5	5	5



**Fig. 1** “Front” (a) and “back” (b) sides of brick

**Table 2** Mean mechanical properties of solid clay bricks

Property	No. of specimens	Value (N/mm <sup>2</sup> )
Compressive strength	7	19.76 (2.5 %)
Flexural strength	7	3.66 (4.3 %)
Splitting tensile strength	7	2.46 (11.4 %)
Direct tensile strength on X dir. (120 mm)	6	1.76 (50 %)
Direct tensile strength on Z dir. (55 mm)	3	1.49 (27 %)
Elastic modulus	6	5,756.00 (5.2 %)

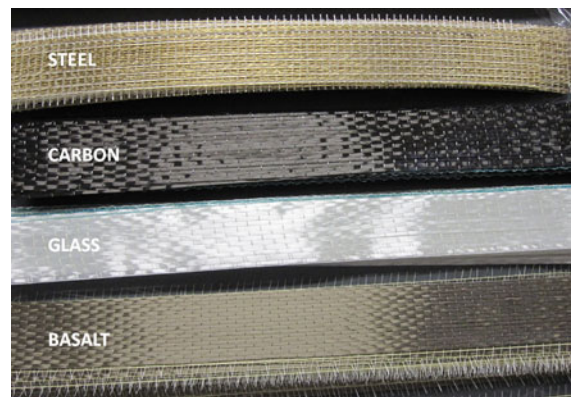
bending tests on the whole unit according to UNI 11128 [73], compression tests according to EN 772-1 [36] and splitting tensile tests according to UNI 8942-3 [75], on each of the two portions of bricks obtained from flexural failure, were carried out. Elastic moduli were measured according to UNI 6556 [74] on samples extracted after splitting tests. The surface of the half-brick specimens used for compression tests were smoothed by mechanical abrasion.

The resulting mean properties are listed in Table 2, which also shows the coefficient of variation (CoV) in brackets.

This type of brick has a pull-off strength of 1.03 N/mm<sup>2</sup> (28 specimens, CoV 11,7 %) evaluated according to ASTM C1583 [12], as reported in Panizza et al. [62].

### 2.1.2 Reinforcing system: properties of composites and resins

Four composite materials comprising glass, basalt, carbon and steel fibers, in the form of unidirectional



**Fig. 2** Fiber fabrics used in experiments

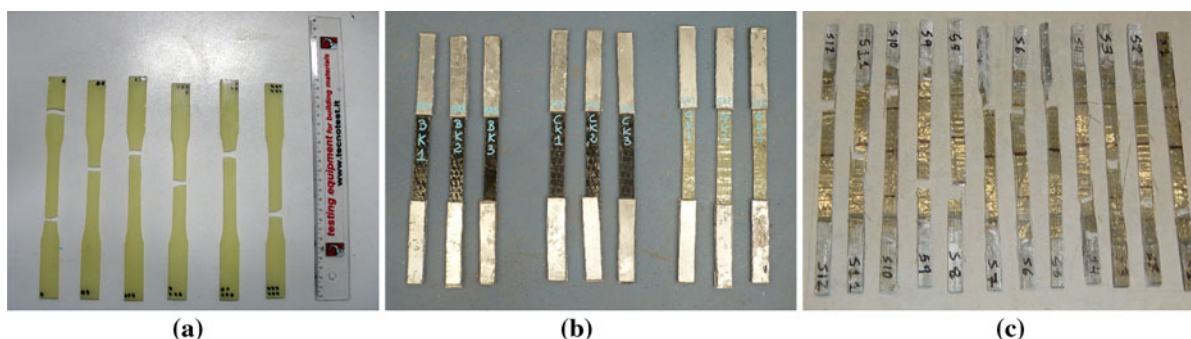
sheets (Fig. 2) were used. They were EB to the bricks with a wet lay-up system, consisting of the application of an epoxy primer on the brick substrate, followed by an epoxy resin and a single sheet of fibers oriented along the length of the brick. The materials used are listed in Table 3. The same product was used as primer for both FRP and SRP specimens; instead, the saturant HM constituted the epoxy resin for FRPs (i.e., glass, basalt and carbon), and the thixotropic saturant HMT was used for steel fibers. All reinforcing materials and systems were provided by FIDIA Technical Global Service, Perugia (Italy).

From technical sheets, the weight of dry sheets before impregnation were as follows: 320 g/m<sup>2</sup> for GFRP, 396 g/m<sup>2</sup> for BFRP, 320 g/m<sup>2</sup> for CFRP, and 1,800 g/m<sup>2</sup> for SRP.

Before performing the shear tests, 27 dog bone specimens of the primer and of both epoxy and thixotropic resins were prepared and tested under tension at UMINHO (specimen length × width ×

**Table 3** Average mechanical properties of primer, resin and fiber specimens tested under tension

Material	No. of specimens	Tensile strength (N/mm <sup>2</sup> )	Young's modulus (N/mm <sup>2</sup> )	Strain at peak load
PRIMER	9	52.6 (7 %)	2,176 (8 %)	3.59 (10 %)
SATURANT HM	9	32.7 (8 %)	1,308 (10 %)	3.77 (6 %)
SATURANT HMT	9	32.9 (8 %)	1,605 (5 %)	3.13 (5 %)
GLASS UNIDIR 300 HT73	21	1,310 (13 %)	84,251 (10 %)	1.69 (15 %)
BASALT UNIDIR 400 C95	21	1,673 (11 %)	88,397 (4 %)	1.96 (12 %)
CARBON UNIDIR 320 HT240	21	2,735 (10 %)	233,861 (5 %)	1.26 (11 %)
STEEL 3X2-B 12-12-500	18	2,997 (7 %)	195,054 (5 %)	1.74 (14 %)

**Fig. 3** Specimens of resins (UNIPD) (a), basalt, carbon and glass (UNIRM3) (b) and steel (UMINHO) (c)

thickness  $185 \times 10 \times 4 \text{ mm}^3$ ) and UNIPD ( $215 \times 13 \times 4 \text{ mm}^3$ ). Likewise, 81 rectangular specimens of impregnated fibers were prepared and tested under tensile loading at UMINHO (single strips of  $400 \times 15 \text{ mm}^2$ ), UNIPD (single strips of  $500 \times 50 \text{ mm}^2$ ) and UNIRM3 (single strips of  $430 \times 60 \text{ mm}^2$ ; and, for glass, carbon and basalt, also three-layer strip specimens of  $300 \times 20 \text{ mm}^2$ ) [2, 13, 66]. Some specimens after testing are shown in Fig. 3. Results are listed in Table 3 in terms of average values, with the CoV in brackets.

It should be noted that the main aim of this phase was to estimate Young's modulus, in order to compute

parameters for analyses (bond stresses, slips, and fracture energy). Therefore, as each laboratory used their available test set-ups, including in-house clamping devices, small inaccuracies may have led to a slight underestimation of tensile strength values, particularly for glass and carbon fibers, mainly due to stress concentration close to the fixing devices.

The elastic modulus was measured by means of a clip gauge and calculated in the range from 30 % to 60 % of maximum load, owing to the linear behavior of the materials almost up to peak load.

From the experimental values listed in Table 3 and in view of the equivalent thickness (weight of fabric per unit area divided by fiber density) of the sheets, the average maximum tensile load for 50-mm wide impregnated fibers, used in the debonding tests described in the next sections, are shown in Table 4.

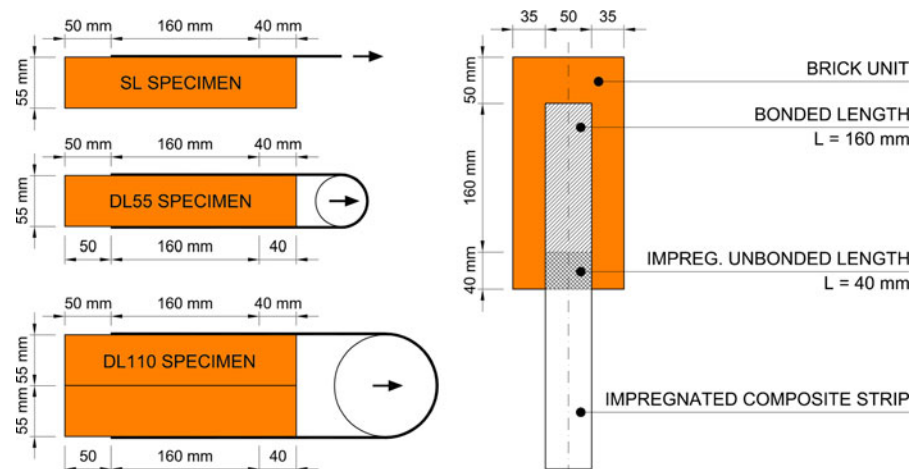
**Table 4** Average peak tensile load computed on 50-mm wide strips of composites

Composite material	Equivalent thickness (mm)	Tensile peak load (N)
GLASS	0.120	7,860
BASALT	0.140	11,795
CARBON	0.170	23,248
STEEL	0.231	34,597

## 2.2 Preparation of specimens

Three groups of specimens, SL, DL55 and DL110, were prepared according to single-lap, double-lap 55

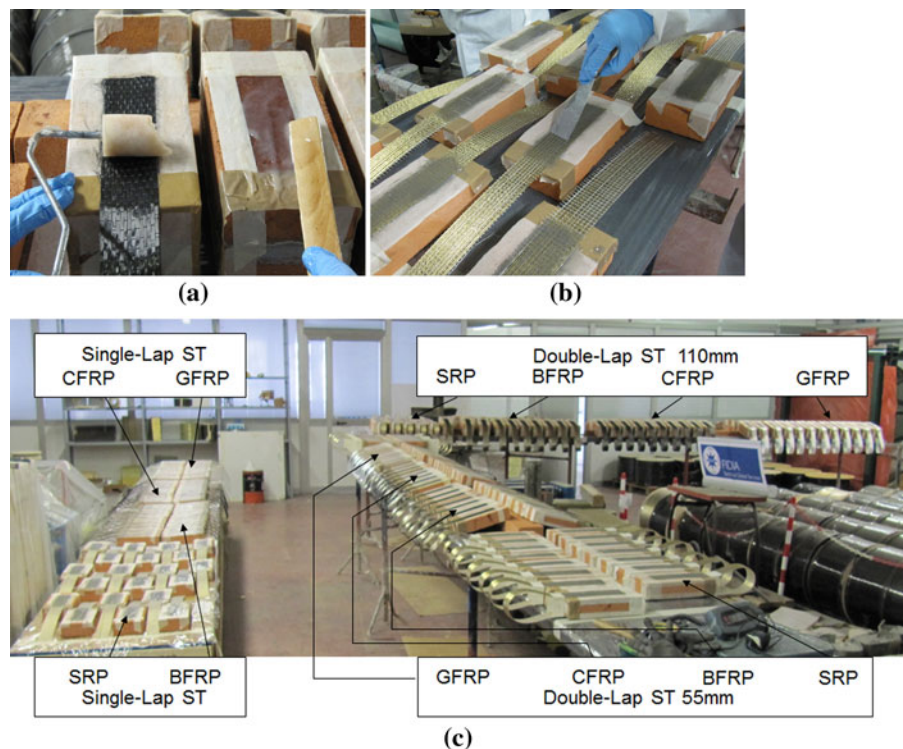
**Fig. 4** Geometry of specimens for SL, DL55 and DL110 set-ups



and double-lap 110 shear test set-ups, respectively (Fig. 4). The reference solid brick as substrate, with dimensions  $55 \times 120 \times 250 \text{ mm}^3$ , was used in all cases. In more detail, SL specimens were built by bonding a single strip of reinforcement along the center line of the front of a single clay brick (see

Fig. 1a). For both DL55 and DL110 specimens, the two ends of the reinforcement strip were EB symmetrically on the opposite surfaces of the bricks, creating a U-shape. In particular, in DL55 specimens, the strip was applied to the two surfaces of a single brick, whereas in DL110 specimens the strip was

**Fig. 5** Preparation of specimens: gluing phase of CFRP DL110 (a), SRP SL (b) and general view of all specimens (c)



applied to both front faces, each specimen being assembled by gluing two bricks together.

Fiber fabrics and wire mesh strips 50 mm wide were obtained manually from sheet rolls by common cutters, when not already provided at the proper width by the manufacturer; in particular, for steel fibers, each strip was composed of 24 strands, each with a section of  $0.481 \text{ mm}^2$ . The strips were bonded to the bricks for a length of 160 mm, leaving an unbonded length of 40 mm, in order to minimize edge effects (Fig. 4).

Particular attention was paid to creating specimens with as few defects and irregularities as possible. Specimens were therefore prepared by the same operators in a relatively short period (about one week during December 2009) and in the same place (SGM Laboratory in Perugia, Italy), except for TECN, UMINHO and UPATRAS, to which materials, complete with detailed videos describing preparation and fiber application, were provided.

Reinforcements were installed taking into account indications provided by ACI 440.2R-08 [1] and CNR DT200 [28]. Before application of the composite strips, dust was removed from the surfaces of units with an industrial vacuum cleaner, to ensure proper bonding of the composite system. After isolation with adhesive tape of the portion of brick not to be glued, a first layer of primer was applied with a small paint roller, in order to penetrate and saturate the unit surface; then a layer of epoxy resin was applied and a small paint roller was used to press the strip into position, ensuring uniform impregnation of fibers and allowing any excess of resin to be squeezed out (Fig. 5a). Lastly, any excess resin was spread with a

palette-knife to create an even surface (Fig. 5b). The ensemble of specimens prepared for the RRT is shown in Fig. 5c.

Specimens were then delivered to the laboratories, proper care being taken during transportation. Except for a few cases, the free lengths of the CFRP, GFRP and BFRP strips were finally impregnated with epoxy resin by each laboratory, in order to guarantee even distribution of tensile forces within the strip during the loading phase.

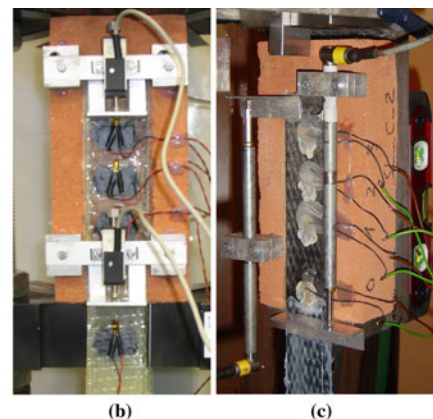
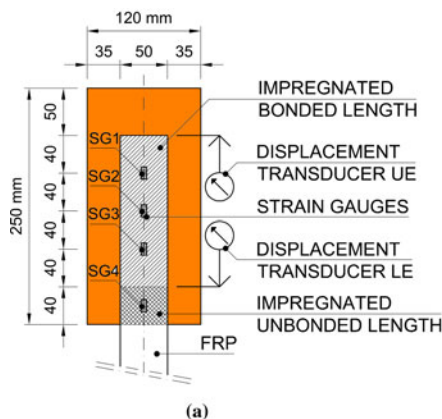
## 2.3 Test set-ups

### 2.3.1 Lay-out of strain and displacement transducers

The main measurement system consisted of four strain gauges placed along the composite strips. When possible, two extra linear transducers were also applied. The strain gauges recorded local deformations along the centerline of the strips, and the transducers recorded the displacement of their loaded (LE) and unloaded (UE) ends. The same type of strain gauge (HBM 1-LY18-6/120) was used for all specimens, and electrical quarter-bridge circuits, compensated for thermal effects by dummy strain gauges, were also used. The displacement transducers differed among laboratories, and the various types included inductive sensors, potentiometers, linear variable differential transducer sensors, and digital indicators.

The general measurement scheme is shown in Fig. 6. The instrumentation lay-out for DL tests was repeated on both sides of the specimens.

**Fig. 6** General lay-out for strain gauge and displacement transducer patterns (a); example of instrumentation applied to DL (UNIPD) (b) and SL (CUT) (c) specimens



### 2.3.2 Single-lap experimental set-ups

SLSTs were performed in seven laboratories, i.e., UNIPD, CUT, UPATRAS, UNIRM3, TECN, UMINHO and UNILE. Various test apparatuses were designed and built in each institution, mostly allowing easy use of the available universal testing machine. Consequently, the steel devices used to place the specimens were slightly different.

Figure 7 shows pictures of the various test set-ups, and Fig. 8 some examples of schemes.

The steel frame used for the tests performed at UNIRM3, UPATRAS and TECN was designed for use with a universal testing machine, and basically consists of stiffened steel plates welded to form an

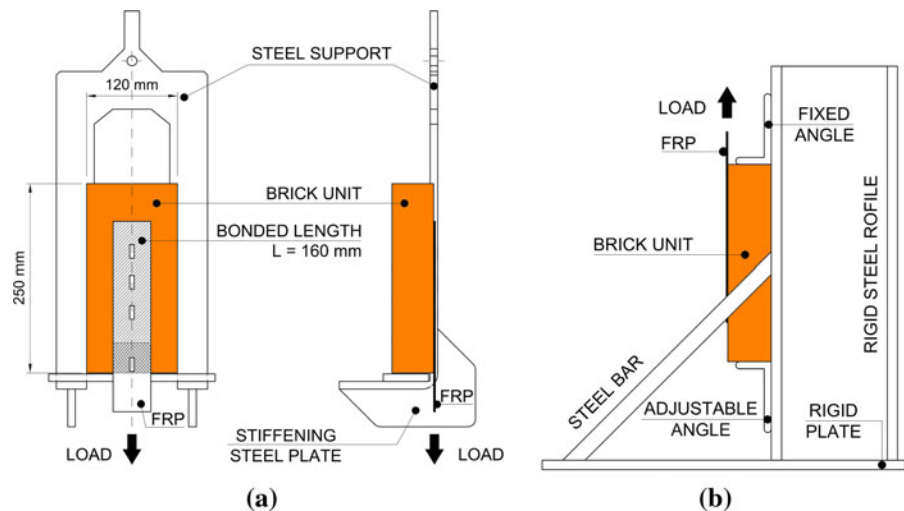
angle of 90° (Fig. 8a). The specimen rests on the bottom of the steel frame and the reinforcement sheet is loaded from below. The frame used at CUT was very similar to that described above, except for the fixing device, which consisted of a stiff steel C-shape frame with a hinge at the top (as at UPATRAS).

Instead, the steel device used at UMINHO and UNILE was designed to be fixed to already available rigid steel testing frames, and the reinforcement sheet was loaded from above. The device is made of a steel profile welded to a rigid plate and stiffened with two diagonal bars (Fig. 8b). The specimen was positioned on the steel device and firmly clamped to it. The system used at UNIPD was similar to the latter, apart from the shape of the steel profiles and the fixing at its base.



**Fig. 7** Test set-ups used at TECN (a), CUT (b), UPATRAS (c), UNIRM3 (d), UNIPD (e), UMINHO (f) and UNILE (g)

**Fig. 8** Schemes of SL set-ups for specimens loaded from below (TECN) (a) or above (UMINHO) (b)



In all cases, the loaded end of the composite strip was glued between two plates (aluminum, steel or GFRP), in order to ensure uniform transmission of load in the clamped area. This system was designed ad hoc by each of the laboratories, by means of bolts, glue, or a combination of both. In particular, UNILE, UNIRM3 and CUT simply glued the end of the reinforcement between two aluminum plates; the same procedure was used at UPATRAS, but with GFRP tabs. A combination of bolts and rapid vinyl-ester resin was used for specimens prepared at UNIPD and UMINHO; TECN chose a clamping system which only had bolts. In the last case, the surface of the steel tabs was roughened, in order to limit slippage between tab and composite strip.

All laboratories, except TECN, used both displacement transducers and strain gauges.

### 2.3.3 Double-lap experimental set-ups

DLSTs were performed in seven laboratories. UNIPD, CUT and UNIPG carried out tests on DL55 specimens, and UNICAS, UNINA, UNICH and UNIRM2 adopted the set-up for DL110 specimens.

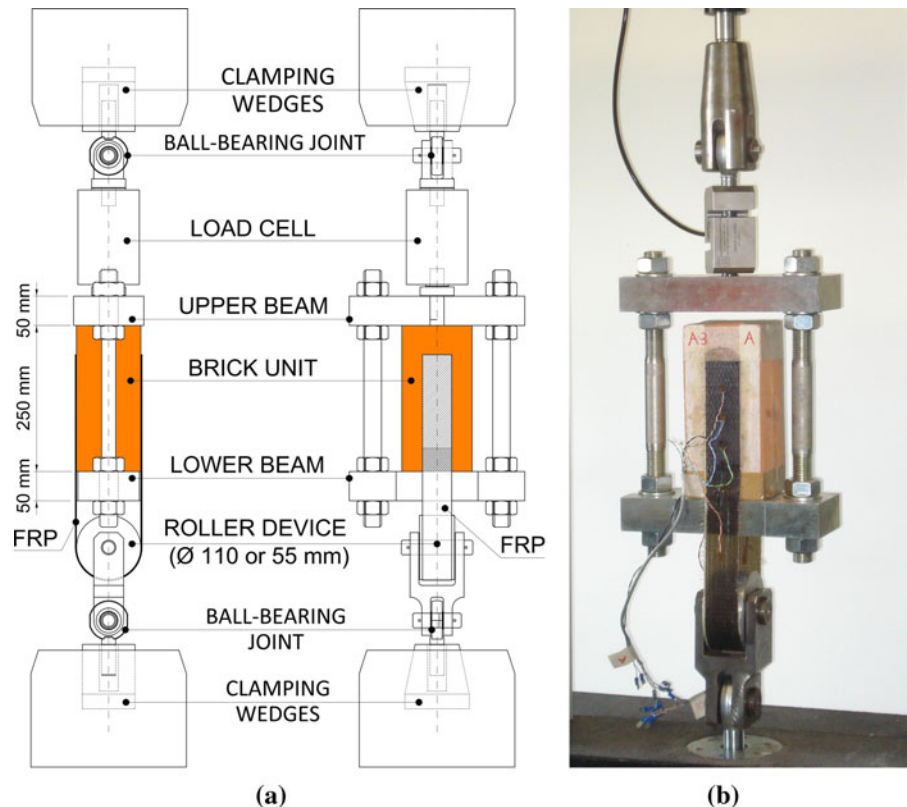
Unlike the SL test set-ups, in which each laboratory designed its own testing frame, the main features of the DL test set-up were discussed among the partners before it was produced. In particular, the following main characteristics were agreed upon: geometrical symmetry of the apparatus, to ensure self-equilibrated set-ups; ball joints at the ends of the steel frame to minimize the effects of any small misalignments; a

roller device to pull the reinforcement to guarantee even loading of the two composite strips; ability of the set-up to accommodate several specimen geometries, in particular, the one- or two-brick thick specimens (types DL55 and DL110, as in Fig. 4); adaptability of the set-up to universal testing machines.

The general scheme consisted of a steel frame composed of two transversal beams connected by two bars. The upper beam was connected to the upper machine head through a load cell, and the specimen rested on the lower beam. The load applied to the strips was intended to be equally divided on both sides of the specimen by a roller device working as a cylindrical hinge (55 or 110 mm in diameter, depending on specimen type). Both connections to the universal machine were made with spherical hinges, to enhance self-alignment. A general view of the whole apparatus and the main steel components for the DL set-up is shown in Fig. 9 and some pictures of the devices during testing in Fig. 10.

The set-ups at UNICAS, UNIPD, UNICH, UNIPG and UNIRM2 exactly resembled that shown in Fig. 9, with some minor adjustments due to different available load cells and universal machines. Some of these laboratories shared the same device. Ball-bearing joints were used at the end connections to the universal machines and frictionless cylindrical hinges were implemented, either by ball bearings or graphite grease around the pin bearing the cylinder, in order to ensure optimal alignment and even loading of the two composite strips (Figs. 9b, 10a, d). At UNINA (Figs. 10c, 11b), the set-up used only one ball-bearing

**Fig. 9** DL set-up: general scheme (a) and typical scheme shared between UNICAS and UNIRM2 (DL110) (b)



joint close to the roller device; at the other end, the bottom plate was clamped directly to the universal machine. The shapes of the bottom plate and the steel frame allowed room for spatial centering and vertical alignment of the specimen, comparable to the general set-up. Lastly, at CUT (Figs. 10b, 11a), brick specimens were fixed inside the steel frame hanging at the top on a ball hinge. At the bottom, fibers were pulled by a non-rotating steel cylinder of 55 mm diameter, capable of allowing small adjustments; friction between cylinder and composite strip was reduced by a series of lubricated plastic leaves and rubber foam inserted at their interface (Fig. 11a).

The laboratories used both displacement transducers and strain gauges, except for UNIRM2, UNICAS and UNIPG; UNICH and UNINA, used only one transducer, positioned at the loaded end.

### 3 Experimental results and analysis

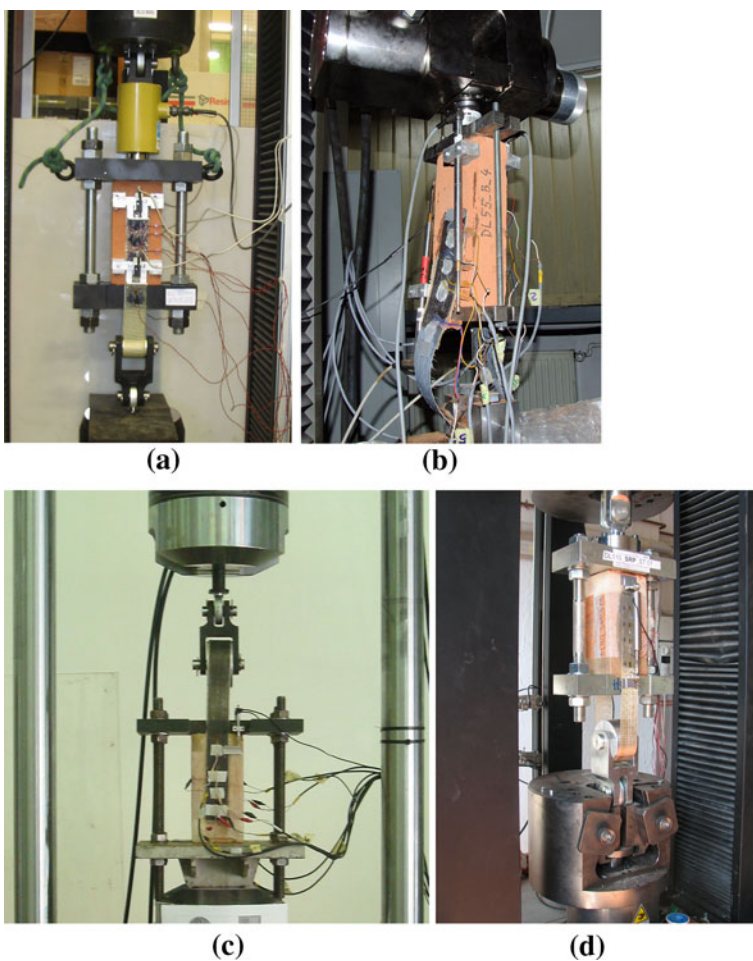
The results of shear tests are discussed for each of the four composite materials and then compared overall.

In the following sections, the outcomes from the various laboratories concerning each composite material are analysed in terms of maximum load, by grouping SL and DL specimens. The influence on failure mode of some aspects concerning set-ups and features of specimens are also discussed. Additional information on the debonding process is also given, through a selection of significant  $\tau$ -slip curves, as well as typical strain profiles along the bonded length. This also allowed estimation of a possible range of variation of the effective transfer length for each of the four materials. Lastly, general behavior during loading is expressed by comparing the significant load versus global displacement curves (measured by transducers, where available).

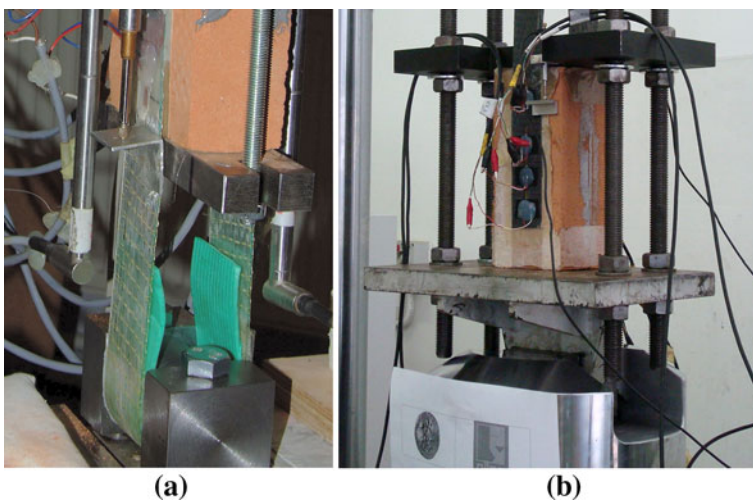
For data analysis, the following assumptions were made for all four materials:

- for DLST, the total load was equally subdivided between the two sides of the brick, and that value was then compared with SLST results;
- to determine the interface  $\tau$ -slip relationship from data recorded by the strain gauges, the average

**Fig. 10** Set-ups used for DL55 tests at UNIPD (a) and CUT (b), set-up for DL110 tests used at UNINA (c), and shared between UNICH and UNIPG (d)



**Fig. 11** Details of non-rotating steel cylinder and slippage system used at CUT (a), and bottom plate clamped directly to universal machine used at UNINA (b)



shear stress between two subsequent strain gauges was estimated by Eq. 1, and the related slipping by Eq. 2 [38, 70]:

$$\tau_{i+1/2} = \frac{E_r A_r (\varepsilon_{i+1} - \varepsilon_i)}{b_r (x_{i+1} - x_i)} \quad (1)$$

where  $A_r$ ,  $b_r$  are the transversal section and width of the reinforcement, respectively;  $E_r$  is the average Young's modulus (as in Table 3), and  $\varepsilon_i$ ,  $x_i$  are the strain and abscissa of the  $i$ -th gauge, respectively.

By assuming perfect bonding between reinforcement and substrate in the very last zone of the anchorage (i.e., no slip occurs at the end of the bonded strip) and by neglecting any deformation of the substrate, slip  $s_{i+1/2}$  in the mid-point between the  $i$ -th and  $(i + 1)$ -th gauges can be estimated by integrating the strain distribution along the reinforcement:

$$s_{i+1/2} = \sum_{k=1}^i (\varepsilon_{k-1} + \varepsilon_k) \frac{\delta_k}{2} + (3\varepsilon_i + \varepsilon_{i+1}) \frac{\delta_{i+1}}{8} \quad (2)$$

where a piecewise uniform strain approximation is adopted and  $\delta_i$  is the distance between the  $(i - 1)$ -th and  $i$ -th gauges.

Common considerations on the four materials concern the failure mode. Indeed, independently of the test set-up, most of the specimens showed similar failure modes, consisting of strip debonding and involving the detachment of a uniform thin layer of the brick along the bonded length. In some cases, minor cracking was observed at the loaded end of the

brick; in other cases, a crack developed in the brick surface near the unloaded side.

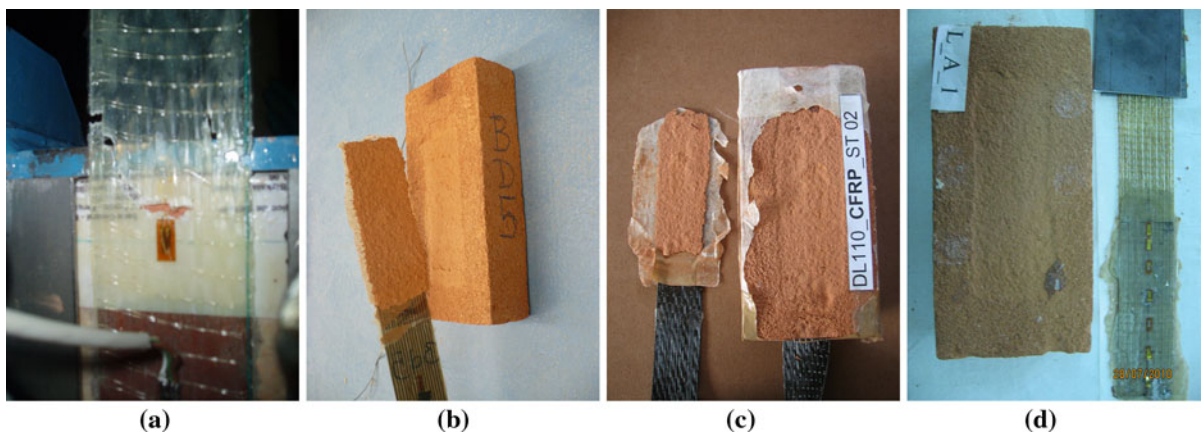
In particular, debonding was observed in 243 tests out of the total of 280 (87 %). The remaining 37 specimens were excluded from some of the comparisons, as they showed anticipated fiber rupture (27 tests, about 10 %) and some anomalies occurred during testing (10 tests, about 3 %). Figure 12 shows some specimens after failure.

For more than 60 % of DL55 specimens which failed due to debonding, the failure occurred in the front of the brick (see Fig. 1), that is, the same side where fibers were applied for the DL110 and SL tests.

Where significant, in the following, specimens excluded from the analysis are marked, as well as specimens in which the free unbonded lengths were not impregnated before testing. In addition, for DL55 specimens, the side of debonding is also marked. The mutual influence of these and other aspects is commented in the following sections and in the overall discussion.

### 3.1 Results on GFRP specimens

Experiments on GFRP comprised 70 tests, half performed in SL test mode and half in DL. Table 5 lists the values of maximum load ( $F_{\max}$ ) obtained from each laboratory, in ascending order for each of the five specimens. The average debonding load ( $F_{\text{mean}}$ ) and the CoV were computed by excluding the results of tests which failed due to fiber rupture or which showed



**Fig. 12** Failure mode in some specimens: tensile failure for GFRP (a), debonding of BFRP (b), CFRP (c) and SRP (d)

**Table 5** Failure loads and average values for GFRP specimens

Laboratory	Set-up	$F_{\max}$ (N)					$F_{\text{mean}}$ (N)	CoV (%)
CUT	SL	3,756	4,404	4,488	4,538	4,612	4,360	7.93
TECN		3,383*	3,660*	3,728*	3,919*	5,667*	–	–
UMINHO		4,160	4,630	4,770	5,050	5,130	4,748	8.14
UNILE		(2,223)*	(2,834)*	3,690	4,057	4,412*	3,874	6.70
UNIPD		4,238	4,240	5,515	5,533	5,613	5,028	14.34
UNIRM3		4,186	4,278	4,312	4,874	5,213	4,573	9.81
UPATRAS		3,881*	4,002*	4,037*	4,312*	5,091	5,091	–
						SL average	4,625	11.87
CUT	DL55	3,675	3,816	4,011	4,044	4,677	4,045	9.49
UNIPD		5,545	5,592	5,651 <sup>a</sup>	6,674*	6,855	5,911	10.68
UNIPG		(–)	(2,552)*	(2,672)*	(3,554)*	(3,620)*	–	–
UNICAS	DL110	4,513*	4,719	4,819	4,910	6,220	5,167	13.67
UNICH		3,835	3,935	4,884	4,984	5,015	4,531	13.08
UNINA		4,178	4,564*	4,642	4,742	5,526	4,772	11.72
UNIRM2		4,505	4,616	4,952	5,280	5,320	4,935	7.54
						DL average	4,850	15.61
						Total average	4,747	14.16

Values in bracket denotes non-impregnated fibers in unbonded lengths

– Results not available, owing to problems in data acquisition

<sup>a</sup> Debonding of back of brick

\* Fiber rupture occurred instead of debonding

problems during the loading phase. Figure 13 shows the direct comparison of debonding loads for the two set-ups.

Out of a total of 70 tests on GFRP specimens, 50 results (71 %) were exploitable for comparative analyses. SL tests provided experimental results which were slightly lower than those of the DL55 and DL110 tests. By excluding those tests which displayed anomalies, the mean value of the failure load of all tests (SL and DL) was 4,747 N, which is 60.4 % of the mean ultimate tensile load of the single strip (7,860 N; see Table 4) (exploitation of fiber strength was 59 % for SL and 62 % for DL tests, including both 55 and 110 types). The CoV of the debonding load was about 14 %. However, it may be noted that the average loads are not very scattered and that the two kinds of specimen have nearly the same failure load (SL 4,625 N and DL 4,850 N, only 5 % of variation); DL110 tests showed results slightly lower (4 %) but slightly more scattered than DL55.

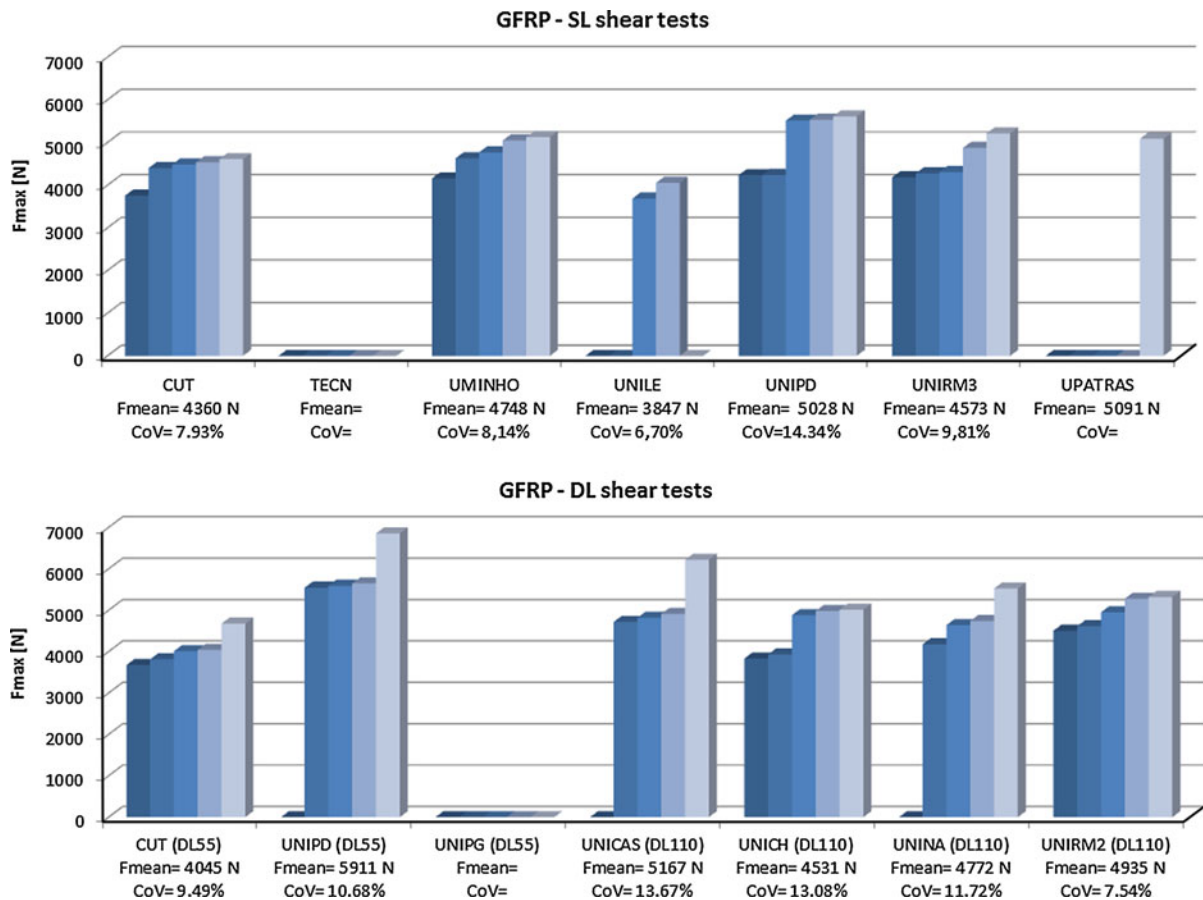
The mean failure load obtained by individual laboratories varied from 3,874 to 5,911 N (CoV 7–14 %); the corresponding mean reinforcement strain at maximum load, measured at failure, was

$0.84 \times 10^{-2}$  (CoV 19.61 %) and was about 50 % of the average ultimate strain of the glass strip ( $1.69 \times 10^{-2}$ ).

According to the test results, the fractile 5 % of the distribution of the debonding load was 3,783 N for the 50-mm wide strip adopted.

To point out the effect of any additional restraints due to some specificities of the DL set-ups used at CUT and UNINA (see Fig. 11), the average values can also be computed without the results obtained at these laboratories. Average loads of 5,091 N (CoV 14.33 %) and 4,830 N (CoV 13.84 %) were obtained by considering all the other more similar testing machines, for DL tests and for the total of DL and SL tests, respectively. The lack of hinges at the base of the system may have caused a reduction of about 5 or 2 % of the average maximum loads, computed for only DL or all tests, respectively.

For all specimens with non-impregnated unbonded strips portions (both for SL and DL tests) fiber rupture was observed before any debonding could occur. Nonetheless, fiber rupture also occurred with the impregnated GFRP strips. Indeed, although only 10 %



**Fig. 13** Glass reinforcement: maximum shear loads for SL and DL tests

of specimens were non-impregnated before testing, more than 27 % (19 specimens) of the total number failed, due to fiber rupture, thus highlighting the specific brittleness of the GFRP and the particular care needed in the impregnation of strips during specimen preparation. Aspects such as the uneven thickness of the impregnating agent, or the lack of fiber straightness during and after impregnation, may have caused uneven distribution of the load among the fibers in the strip and premature rupture of the GFRP composite.

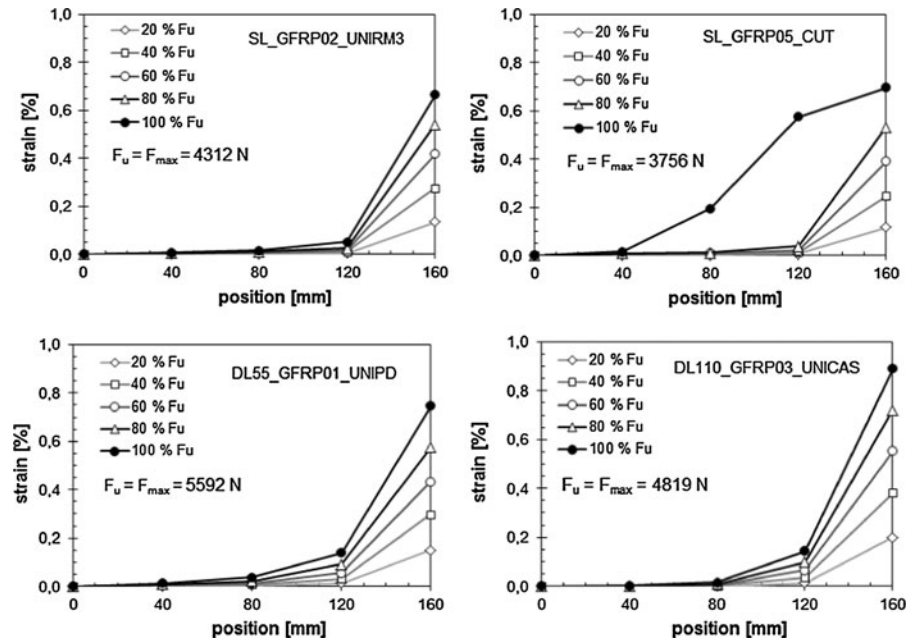
Results showed that the SL and DL55 test set-ups affect fiber rupture more than DL110: 34 % (12 specimens), 33 % (5) and 10 % (2) of tests were excluded due to this effect, respectively. For SL specimens, a too thin or too thick layer of resin at the clamping device may have been responsible for slippage of the loaded end of the strip in the test machine. Another factor may have been the types of clamping device used (mechanical or hydraulic/pneumatic). In addition, some

SL set-ups may have suffered from non-uniform load distribution over the reinforcement width.

Some representative axial strain profiles along the bonded length (160 mm) obtained during testing are shown in Fig. 14. The strain gauges applied to the bonded length were: SG1 (40 mm position), SG2 (80 mm) and SG3 (120 mm). SG4 is positioned out of the bonded length (160 mm), assuming uniform strain profile throughout the unbonded portion (see Fig. 6). Profiles were recorded for load levels between 0.2 and 1 of the failure load, as shown in the plots. Except for some of the laboratories, which had some problems with strain gauge recordings, good agreement in strain profiles was obtained, independently of test set-up.

The curves remained almost exponential until the debonding load was reached. In all cases, the last strain gauge on the bonded length, SG3, was activated during the loading process; slight deformations were also recorded in SG2 for some of the tests, close to the

**Fig. 14** Glass reinforcement: strain profiles for SL and DL tests

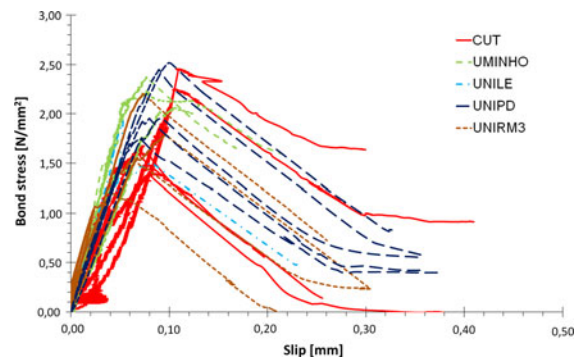


debonding load, as shown in Fig. 14. On the basis of these results, it did not seem necessary to test higher bond lengths for glass fibers in order to analyse effective transfer stress mechanisms, and the effective transfer length was estimated of around 80 mm.

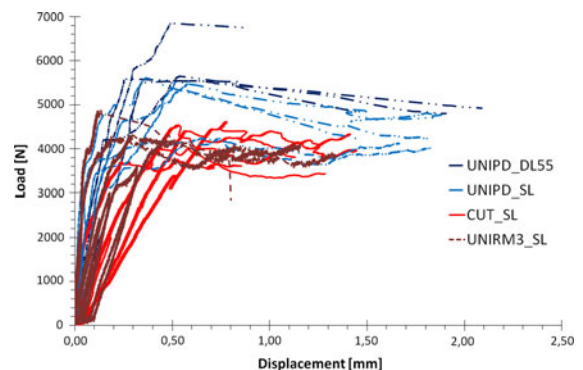
Considering all the results, the maximum strain in unbonded fibers when debonding occurred was about  $0.7 \div 1.2 \times 10^{-2}$ , and the maximum strain corresponding to the failure of GFRP in tension was about  $1.69 \times 10^{-2}$ ; therefore, debonding takes place at about  $41 \div 71$  % of the fiber strength. As the debonding process develops, a rapid increase in the strain recorded by SG3 (it starts working like SG4, positioned in the unbonded region), and the succeeding bond region (SG2) is activated (see Fig. 14, CUT).

Integrating the strain profiles recorded by pairs of consecutive strain gauges, neglecting deformation in the substrate, the GFRP-to-brick bond stress versus slip curve was derived as shown in Fig. 15. Apart from a certain scatter, which is typical in these cases, the experiments provided a shear stress–slip interface relationship with a quasi-linear ascending part, ending at a maximum shear stress value ranging between 1.2 and  $2.5 \text{ N/mm}^2$  and a corresponding slip of about 0.09 mm, followed by a softening part which, from extrapolation of experimental results, ends (at zero shear stress) between 0.20 and 0.50 mm.

Lastly, the debonding load versus the global displacement curve is plotted for some selected cases



**Fig. 15** Glass reinforcement: some representative shear stress–slip curves



**Fig. 16** Glass reinforcement: load–displacement curves

in Fig. 16. Global slip was measured by displacement transducers, when available. An initial linear response up to a slip of about 0.2–0.3 mm was observed, followed by an almost horizontal part, with average maximum load of about 4,700 N. This can add information to the estimate of the effective transfer length, as the observed behavior indicates that the required bond length is shorter than the adopted bond length, whereas elastic-fragile overall behavior would mean that the required bond length would be higher than the adopted one.

Generally, in the laboratories where displacement transducers were also used, the highest displacement values at maximum load were measured on the side where the failure occurred. In particular, the average maximum displacements were 0.87, 0.76 and 0.83 mm for DL110, DL55 and SL specimens, respectively.

### 3.2 Results on BFRP specimens

Experiments on BFRP comprised 70 tests, half performed in SL test mode and half in DL. Table 6 lists the values of maximum load ( $F_{\max}$ ) obtained from each laboratory, in ascending order for each of the five

specimens. The average debonding load ( $F_{\text{mean}}$ ) and CoV were computed by excluding the results of tests which failed due to fiber rupture or which showed problems during the loading phase. Figure 17 shows the direct comparison of debonding loads for the two set-ups.

Out of a total of 70 tests on BFRP specimens, 62 results (89 %) were exploitable for comparative analyses. SL tests provided less scattered experimental results than those of the DL55 and DL110 tests. By excluding those tests which displayed anomalies, the mean value of the failure load of all tests (SL and DL) was 5,261 N, which is 44.6 % of the mean ultimate tensile load of the single strip (11,795 N; see Table 4) (exploitation of fiber strength was 45 % for SL tests, 42 % for DL55 and 46 % for DL110). The CoV of the debonding load was 16.9 %. However, note that the average loads are not very scattered and that the two kinds of specimen have nearly the same failure load (SL 5,313 N and DL 5,219 N, only 2 % of variation); DL110 tests showed results slightly higher (9 %) and more scattered than DL55.

The mean failure load obtained by individual laboratories varied from 4,442 to 6,094 N (CoV

**Table 6** Failure loads and average values for BFRP specimens

Laboratory	Set-up	$F_{\max}$ (N)					$F_{\text{mean}}$ (N)	CoV (%)
CUT	SL	4,199	4,305	4,423	4,518	5,598	4,609	12.28
TECN		4,996*	5,140	5,668	5,742	6,345	5,724	8.62
UMINHO		5,930	6,000	6,050	6,240	6,250	6,094	2.37
UNILE		(3,201)*	3,813	4,667	4,791	5,403	4,669	14.03
UNIPD		4,828	5,291	5,695	5,868	5,900	5,516	8.24
UNIRM3		4,364	5,043	5,104	5,220	6,366	5,219	13.85
UPATRAS		(4,907)*	(5,097)*	(5,278)*	(5,562)*	(5,631)*	–	–
						SL average	5,313	13.75
CUT	DL55	4,692	4,875	5,004	5,427	5,979 <sup>a</sup>	5,195	9.91
UNIPD		3,887	4,681	5,574	5,878 <sup>a</sup>	6,247	5,253	18.25
UNIPG		3,070	3,630	4,682	5,268 <sup>a</sup>	5,558 <sup>a</sup>	4,442	23.95
UNICAS	DL110	4,854	5,200	5,665	5,685	7,691	5,819	18.94
UNICH		3,342	3,860	4,347	5,390	5,896	4,567	23.20
UNINA		3,054*	4,424	4,678	5,818	6,366	5,322	17.35
UNIRM2		4,486	6,085	6,295	6,345	6,560	5,954	14.07
						DL average	5,219	19.36
						Total average	5,261	16.90

Values in bracket denotes non-impregnated fibers in unbonded lengths

<sup>a</sup> Debonding of back of brick

\* Fiber rupture occurred instead of debonding

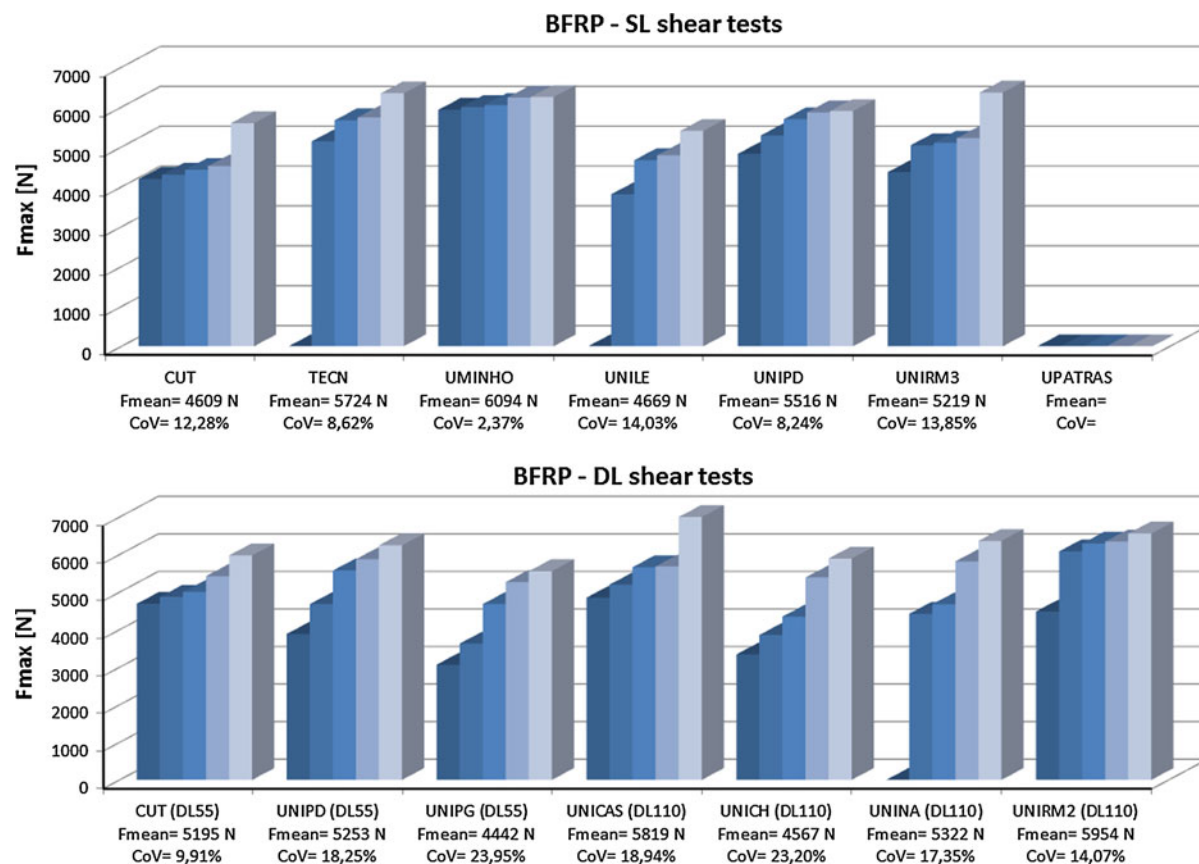


Fig. 17 Basalt reinforcement: maximum shear loads for SL and DL tests

2.4–24 %); the corresponding mean reinforcement strain at maximum load, measured at failure, was  $0.72 \times 10^{-2}$  (CoV 34.87 %) and was 36.9 % of the average ultimate strain of the basalt strip ( $1.96 \times 10^{-2}$ ).

According to the test results, the fractile 5 % of the distribution of the debonding load was 3,815 N for the 50-mm wide strip adopted.

Excluding the values of DL results from CUT and UNINA, average loads of 5,207 N (CoV 20.82 %) and 5,263 N (CoV 17 %) were obtained for DL tests and for the total of DL and SL tests, respectively. These results show that the lack of hinges at the base of the system does not significantly influence the average maximum load.

As for GFRP, all BFRP specimens with non-impregnated unbonded portions of the strips (6 out of the total of 70) failed, due to fiber rupture. Two further specimens, one for SL and one for DL (110), although impregnated, also failed due to fiber rupture. This result may be considered better than that for GFRP, in

which fiber rupture occurred in several specimens impregnated in their unbonded portions.

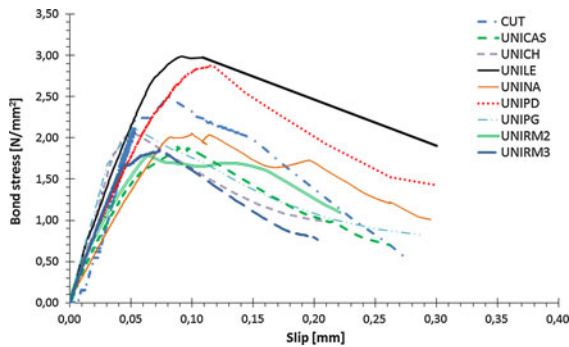
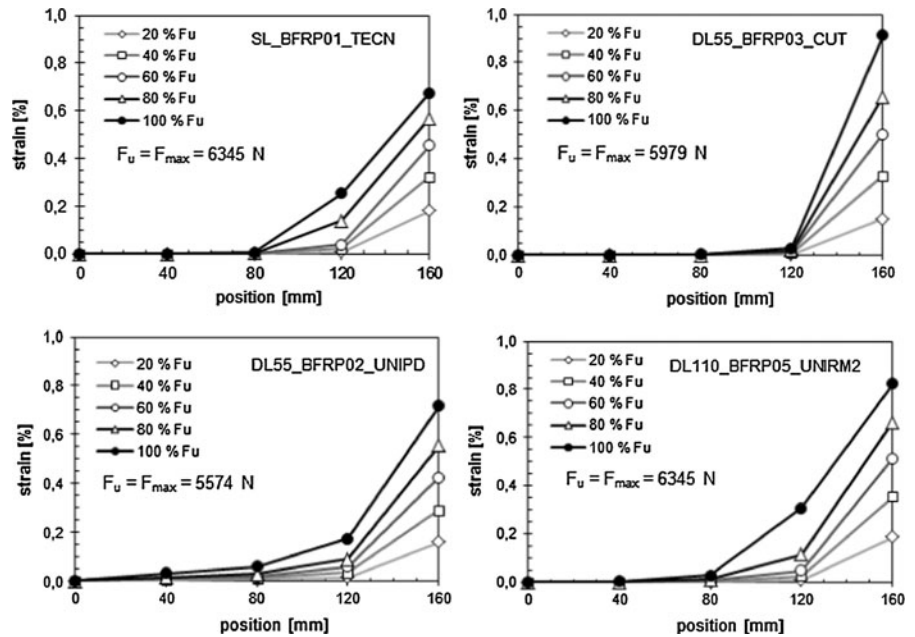
Some axial strain profiles along the bonded length obtained during testing are shown in Fig. 18.

In most cases, SG1 and SG2 measured very low values of strain. In SG3, strain often suddenly increased, at a load level of 60–70 % of maximum load. On the basis of these results, the effective transfer length of the reinforcing system seems to be around 80 mm.

However, it must be noted that some differences among the results obtained by the various laboratories were observed. Indeed, in some cases (e.g., Fig. 18, CUT), the strain profiles showed a bonding length of about 40 mm, involving only the first strain gauge applied on the bonded length. In other cases (e.g., Fig. 18, UNIPD), even SG1 recorded non-zero strain values at debonding.

Integrating the strain profiles recorded by pairs of consecutive strain gauges, neglecting the deformation

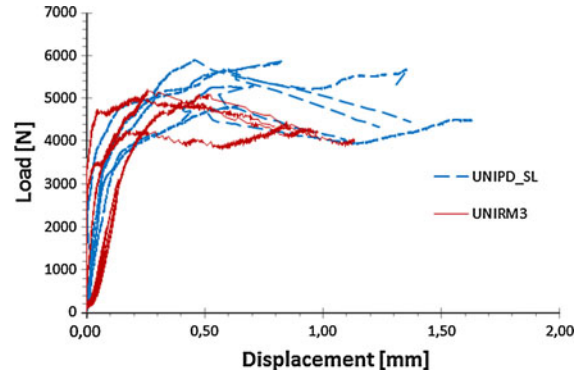
**Fig. 18** Basalt reinforcement: strain profiles for SL and DL tests



**Fig. 19** Basalt reinforcement: some representative shear stress–slip curves

in the substrate, the BFRP-to-brick bond stress versus slip curve was derived as shown in Fig. 19. The tests provided a shear stress–slip interface relationship with a quasi-linear ascending branch (up to about 90 % of peak load) ending with a maximum shear stress value varying between 1.8 and 3.0 N/mm<sup>2</sup> and a corresponding slip between 0.06 and 0.12 mm, followed by a softening, almost linear part which, based on experimental values, reached zero shear stress between 0.30 and 0.60 mm. The mean value of the peak stress and of the slip at the end of the softening part were 2.24 MPa and about 0.40 mm, respectively.

Lastly, the curve of the load versus displacement at the end of the bonded strip is plotted for some selected



**Fig. 20** Basalt reinforcement: load–displacement curves

cases in Fig. 20. A nearly elastic part followed by an almost flat one was observed. Peak loads varied between 4,199 and 5,900 N, with a slip corresponding to peak loads of 0.16–0.60 mm.

### 3.3 Results on CFRP specimens

Experiments on CFRP comprised 70 tests, half performed in SL test mode and half in DL. Table 7 lists the values of maximum load ( $F_{max}$ ) obtained from each laboratory, in ascending order for each of the five specimens. The average debonding load ( $F_{mean}$ ) and the CoV were computed by excluding the results of tests which showed problems during the loading

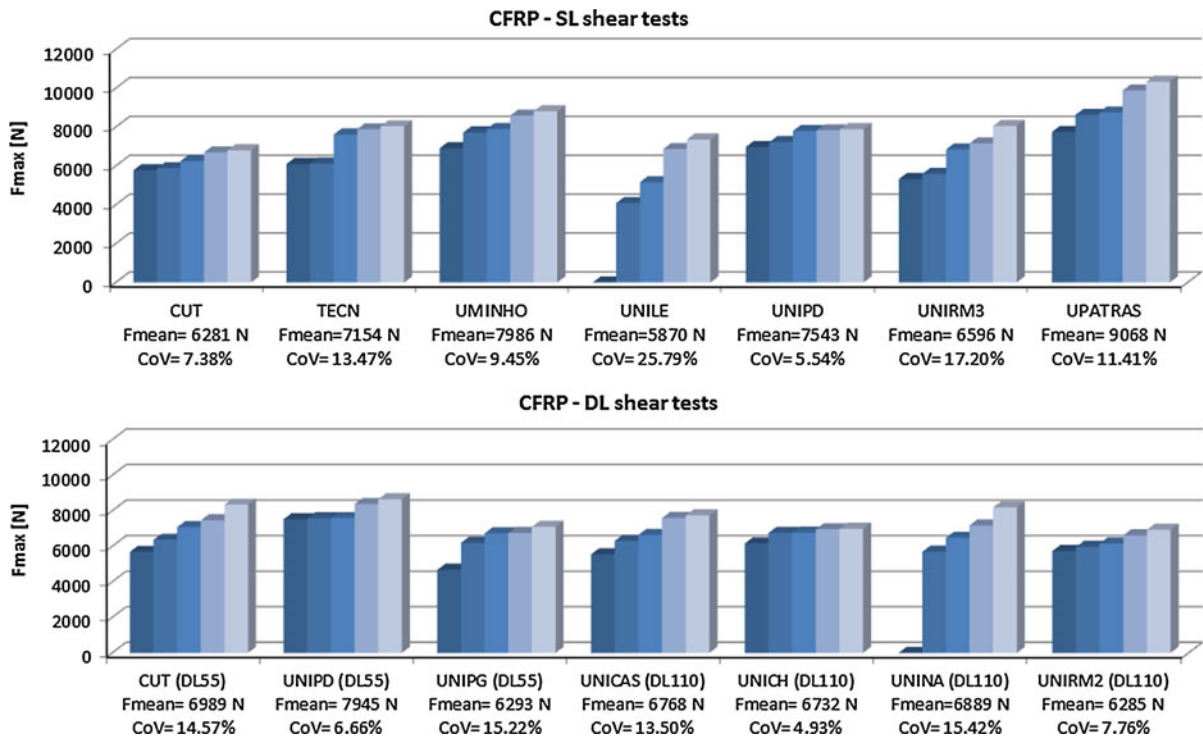
**Table 7** Failure loads and average values for CFRP specimens

Laboratory	Set-up	$F_{\max}$ (N)					$F_{\text{mean}}$ (N)	CoV (%)
CUT	SL	5,778	5,878	6,251	6,691	6,806	6,281	7.38
TECN		6,098	6,126	7,615	7,889	8,042	7,154	13.47
UMINHO		6,910	7,720	7,900	8,590	8,810	7,986	9.45
UNILE		–	4,094	5,158	6,870	7,359	5,870	25.79
UNIPD		6,971	7,223	7,796	7,833	7,891	7,543	5.54
UNIRM3		5,320	5,595	6,845	7,163	8,057	6,596	17.20
UPATRAS		(7,750)	(8,630)	(8,750)	(9,880)	(10,330)	9,068	11.41
						SL average	7,254	18.28
CUT	DL55	5,688	6,372	7,077	7,464 <sup>a</sup>	8,346	6,989	14.57
UNIPD		7,525	7,581	7,590	8,382	8,649	7,945	6.66
UNIPG		4,680 <sup>a</sup>	6,200 <sup>a</sup>	6,732	6,755 <sup>a</sup>	7,100 <sup>a</sup>	6,293	15.22
UNICAS	DL110	5,550	6,310	6,645	7,595	7,740	6,768	13.50
UNICH		6,171	6,758	6,771	6,964	6,995	6,732	4.93
UNINA		–	5,694	6,484	7,180	8,196	6,889	15.42
UNIRM2		5,735	5,970	6,165	6,620	6,935	6,285	7.76
						DL average	6,842	13.07
						Total average	7,048	16.20

Values in bracket denotes non-impregnated fibers in unbonded lengths

– Results not available, owing to problems in data acquisition

<sup>a</sup> Debonding of back of brick

**Fig. 21** Carbon reinforcement: maximum shear loads for SL and DL tests

phase. Figure 21 shows the direct comparison of debonding loads for the two set-ups.

Out of a total of 70 tests on carbon, 68 results (97 %) were exploitable for comparative analyses; only one SL and one DL110 results were not available, due to problems during data acquisition. Although non-impregnated free ends were adopted in some specimens (six SL cases), no fiber rupture was observed. The SL tests showed load values slightly higher than those of the DL55 and DL110 tests. Excluding those tests which displayed anomalies, the mean value of the failure load of all tests (SL and DL) was 7,048 N, which is 30.3 % of the mean ultimate tensile load of the single strip (23,248 N; see Table 4) (exploitation of fiber strength was 31 % for SL tests and 29 % for DL tests, including both 55 and 110 types). The CoV of the debonding load was about 16 %. However, note that the average loads are not very scattered and that the two kinds of specimen have nearly the same failure load (SL 7,254 N and DL 6,842 N, 6 % of variation); likewise, DL110 tests showed results slightly higher (6 %) but slightly less scattered than DL55.

The mean failure load obtained by individual laboratories varied from 5,870 to 9,068 N (CoV 5–26 %); the corresponding mean reinforcement strain at maximum load, measured by SG4 at failure, was  $3.8 \times 10^{-3}$  (CoV 43.10 %) in SL and  $4.3 \times 10^{-3}$  (CoV 34.12 %) in DL (average values of  $4.05 \times 10^{-3}$  and CoV 38.6 %); these values were about 33 % of the average ultimate strain of the carbon strip ( $1.26 \times 10^{-2}$ ).

According to these results, the fractile 5 % of the distribution of the debonding load was 5,400 N for the 50-mm wide strip adopted.

Without taking into account the DL CUT and UNINA results, the DL average ultimate load value was 6,826 N (CoV of 13.23 %) and the total average value was 7,077 N (CoV of 16.67 %). Therefore, also in this case the lack of hinges at the base of the system had a minimum effect (increase by about 0.4 % of average maximum loads).

Some axial strain profiles obtained during testing along the bonded length are shown in Fig. 21. Different results emerged from the various experiments. For SLST, UNILE noted that the analysed bonded length (160 mm) may be shorter than the effective transfer length (at least in two samples). Conversely, CUT observed the activation only of the strain gauges (one or two) closest to the loaded side. In

DLST, no activation of any of the strain gauges at low load levels was detected at UNICH (DL 110) and CUT (DL 55). Conversely, UNICAS and UNINA recorded activation of all strain gauges along the bonded length, even at low load levels (less than 20 or 40 % of failure load). In these last cases, the whole bonded length of the strip was involved in carrying the external load; strain distribution along the bonded length showed that the loaded length was almost comparable to the bonded length, as all three strain gauges on each side were involved. These differing results may be attributed to the sensitivity of the strain gauges at the lower strain levels and interpretations of the data at lower strain levels.

Only in a few cases was the first strain gauge (SG1) activated before debonding, whereas SG2 was almost always activated. Therefore, we may state that the effective transfer length is greater than 80 mm, probably around 120 mm but no higher than 160 mm, as also deducible from the plots of Fig. 22.

The maximum strain in the unbonded strip when debonding occurred was  $3 \div 4 \times 10^{-3}$ .

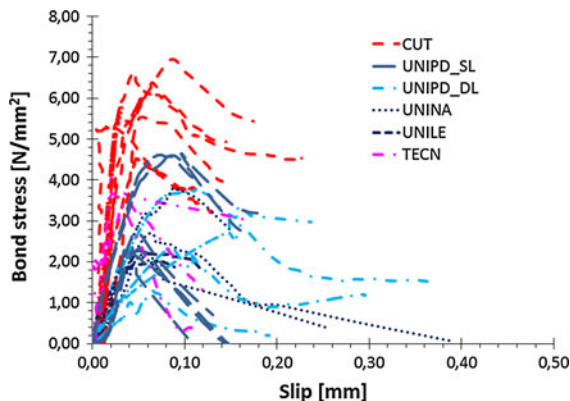
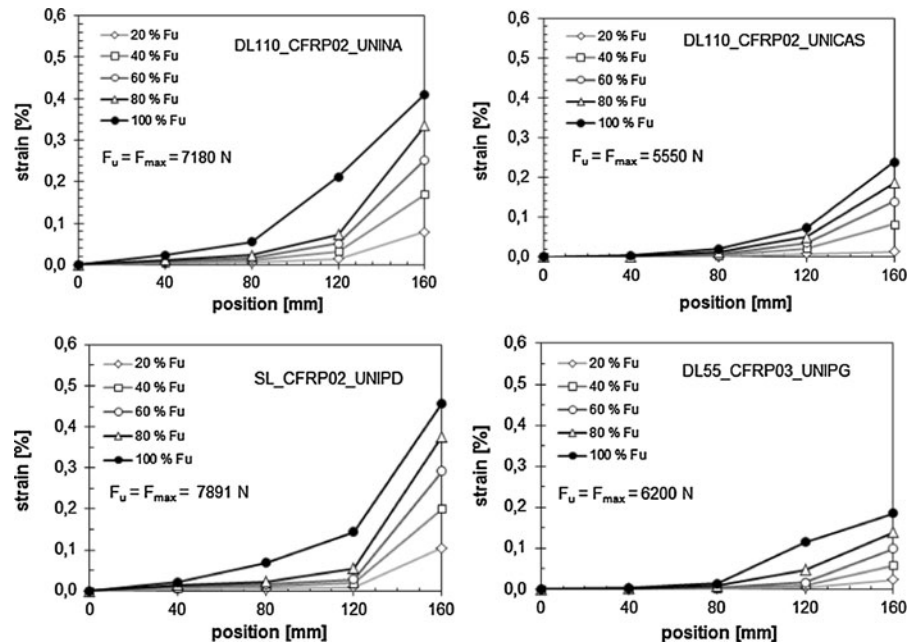
Integrating the strain profiles recorded by the strain gauges, the CFRP-to-brick bond stress versus slip curve was derived as shown in Fig. 23. Apart from a certain scatter, the experiments provided a shear stress–slip interface relationship with a quasi-linear ascending part (up to about 90 % of peak load), mostly ending at a maximum shear stress value of 2.0–4.5 N/mm<sup>2</sup> and a corresponding slip of about 0.07 mm, followed by a softening part. In this case, the experimental results were quite scattered, being the ultimate slip corresponding to zero shear stress between 0.1 and about 0.55 mm.

The load versus displacement curve is plotted in Fig. 24. Piecewise linear overall behavior is obtained with an average maximum load of about 8,000 N. Most specimens showed initial elastic behavior, followed by an almost horizontal part, whereas only one or two cases may be considered as brittle-elastic, thus confirming that the effective transfer length is around 120–160 mm.

### 3.4 Results on SRP specimens

Experiments on SRP comprised 70 tests, half performed in SL test mode and half in DL. Table 8 lists the values of maximum load ( $F_{\max}$ ) obtained from each laboratory, in ascending order for each of the five specimens. The

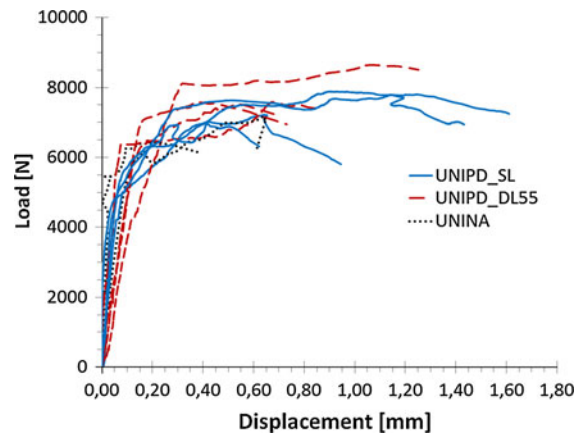
**Fig. 22** Carbon reinforcement: strain profiles for SL and DL tests



**Fig. 23** Carbon reinforcement: some representative shear stress–slip curves

average debonding load ( $F_{\text{mean}}$ ) and the CoV were computed by excluding seven tests (three for SL and four for DL) because the data were either anomalous or not available, due to problems with the data acquisition system. Figure 25 shows the direct comparison of debonding loads for the two set-ups.

Due to the characteristics of the strips, there was no need to impregnate their free portions. SL tests provided load values slightly higher (7.5 %) and less scattered than in DL55 and DL110. The mean value of the debonding load computed among 62 tests was 7,568 N, which is 21.9 % of the mean ultimate tensile load of the



**Fig. 24** Carbon reinforcement: load–displacement curves

single strip (34,597 N; see Table 4) (exploitation of fiber strength was 23 % for SL tests, and 21 % for both DL55 and DL110). The coefficient of variation of the debonding load is about 16 %. DL110 and DL55 tests showed results very close (only 0.2 % of variation), but DL110 provided values more scattered than DL55.

The mean failure load obtained by individual laboratories varied from 5,954 to 8,766 N (CoV 12–14 %); the corresponding mean reinforcement strain at maximum load, measured at failure, was  $3.56 \times 10^{-3}$  (CoV 24.77 %) and was about 20 % of the average ultimate strain of the steel strip ( $1.74 \times 10^{-2}$ ).

**Table 8** Failure loads and average values for SRP specimens

Laboratory	Set-up	$F_{\max}$ (N)					$F_{\text{mean}}$ (N)	CoV (%)
CUT	SL	5,697	6,490	6,739	7,172	7,523	6,724	10.38
TECN		7,104	7,261	7,430	7,854	8,831	7,696	9.01
UMINHO		6,550	7,150	7,750	8,110	9,530	7,818	14.41
UNILE		–	5,280	6,870	7,237	10,380	7,442	28.69
UNIPD		7,143	7,894	8,267	8,367	9,940	8,322	12.31
UNIRM3		–	–	7,670	7,843	8,872	8,129	7.99
UPATRAS		7,175	8,511	8,787	9,614	9,744	8,766	11.79
						SL average	7,837	15.31
CUT	DL55	5,820	6,054 <sup>b</sup>	7,354	7,365 <sup>b</sup>	7,896	6,898	13.16
UNIPD		8,246 <sup>a</sup>	12,199 <sup>a,b</sup>	12,948 <sup>a,b</sup>	13,873 <sup>a,b</sup>	13,980 <sup>a,b</sup>	8,246	–
UNIPG		6,354 <sup>a</sup>	6,652 <sup>a</sup>	6,772 <sup>b</sup>	7,324	8,066 <sup>a</sup>	7,034	9.61
UNICAS	DL110	7,060	7,300	7,570	8,755	9,215	7,980	11.89
UNICH		7,114	7,395	7,987	8,487	9,225	8,042	10.55
UNINA		6,264	6,394	6,656	8,660	10,240	7,643	22.86
UNIRM2		4,486	6,085	6,295	6,345	6,560	5,954	14.07
						DL average	7,290	16.50
						Total average	7,568	16.16

– Results not available, owing to problems in data acquisition

<sup>a</sup> Debonding of back of brick

<sup>b</sup> Anomalous values

According to these test results, the fractile 5 % of the distribution of the debonding load was 5,843 N for the 50-mm wide strip adopted.

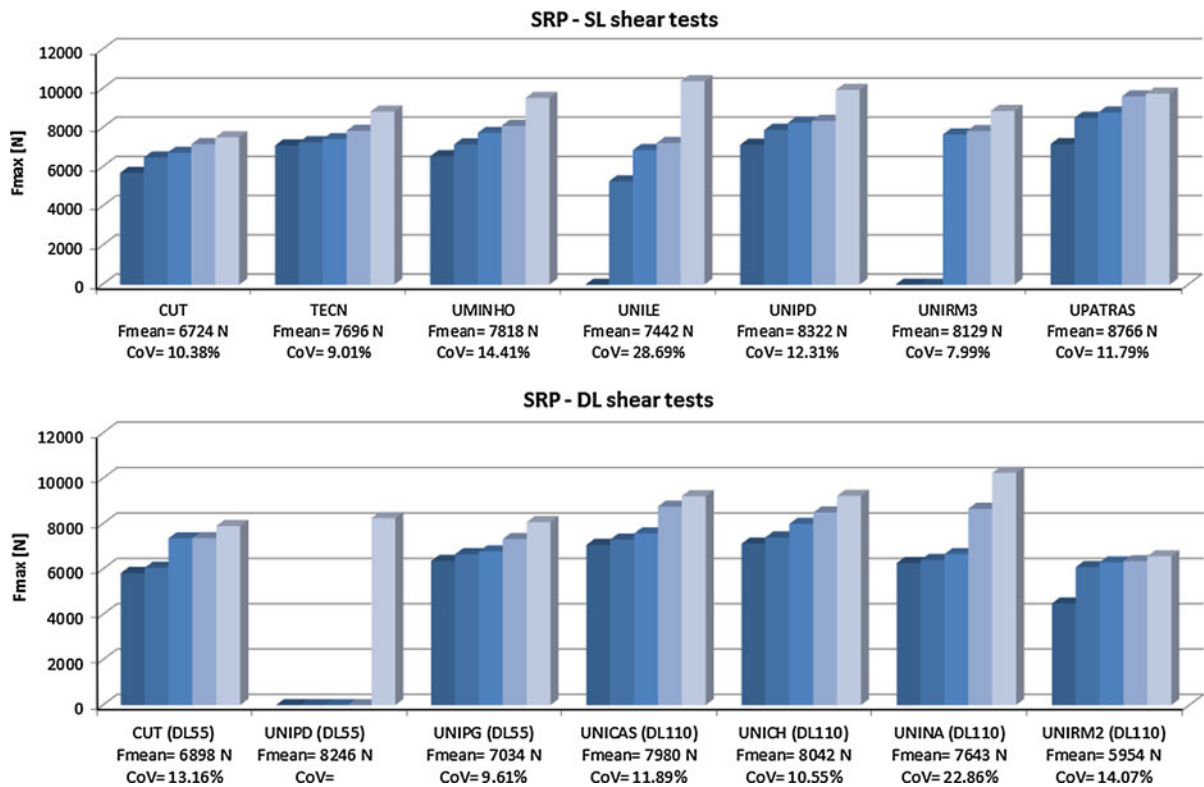
Without taking into account the DL CUT and UNINA results, the DL average value of ultimate load was 7,300 N (CoV of 15.76 %) and the total average value was 7,624 N (CoV of 15.73 %). The lack of hinges at the base of the system only had a minimum effect (increase by about 0.7 % of average maximum loads).

Some axial strain profiles obtained during testing along the bonded length are shown in Fig. 26. Note that the debonding process develops with non-negligible strain in the gauge located near the end of the bonded length (SG1): therefore, the whole bonded length of the strip is involved in carrying the external load, i.e., the effective transfer length is about the same length or longer. Apart from some of the laboratories, which had problems in strain gauge recordings, very good agreement in load profiles was obtained, independently of test set-up. The maximum strain in unbonded strips prior to debonding was about  $3.0\text{--}4.5 \times 10^{-3}$ . The strain profiles along the strips

remained almost exponential until failure and the deformations recorded on the unbonded region near the loaded end (SG4) were always many times higher than those recorded at the end of the bonded region (SG1).

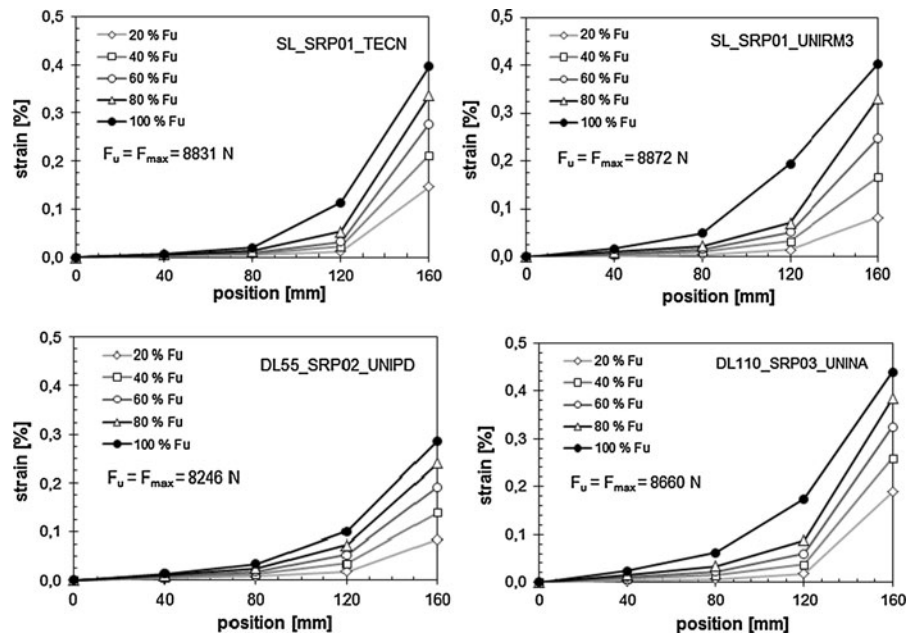
Integrating the strain profiles recorded by the strain gauges, the SRP-to-brick bond stress versus slip curve was derived as shown in Fig. 27. Apart from a certain scatter, the experiments provided an interface shear stress–slip relationship with a linear ascending part ending at a maximum shear stress of 1.5–4.0 N/mm<sup>2</sup> and a corresponding slip of about 0.05 mm, followed by a softening part ending at about 0.10–0.25 mm.

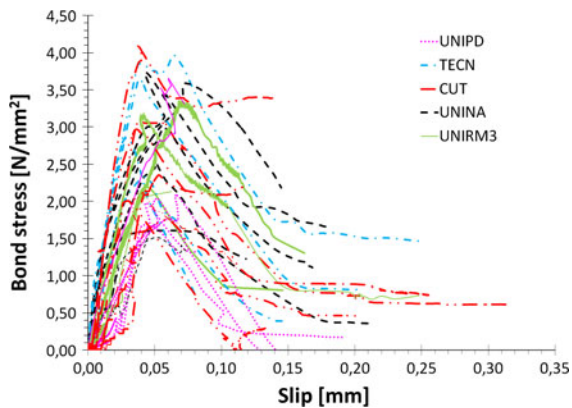
The strain profiles show that the effective transfer length was greater than 120 mm, since in many cases SG1 was activated before failure. Moreover, looking at the load–displacement curves (Fig. 28), obtained by direct measurement of slip recorded with displacement transducers at the end of the bonded strip, brittle failure was recorded in some cases. Therefore, it cannot be excluded that the required transfer length in those cases was greater than the bonded length of the specimens (160 mm). For the average maximum load



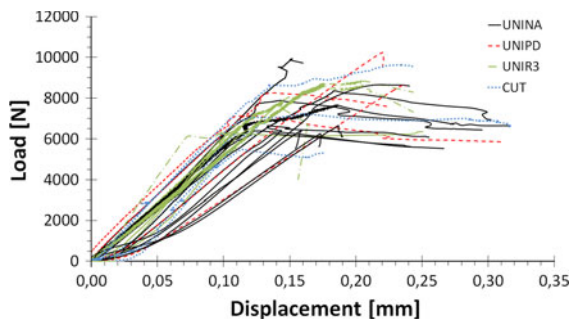
**Fig. 25** Steel reinforcement: maximum shear loads for SL and DL tests

**Fig. 26** Steel reinforcement: strain profiles for SL and DL tests





**Fig. 27** Steel reinforcement: some representative shear stress–slip curves



**Fig. 28** Steel reinforcement: load–displacement curves

(7,568 N), a displacement of about 0.14 mm was recorded.

### 3.5 Overall comparison of results

The results, previously presented in detail for each type of reinforcing system, are compared in this section, allowing some general observations to be made.

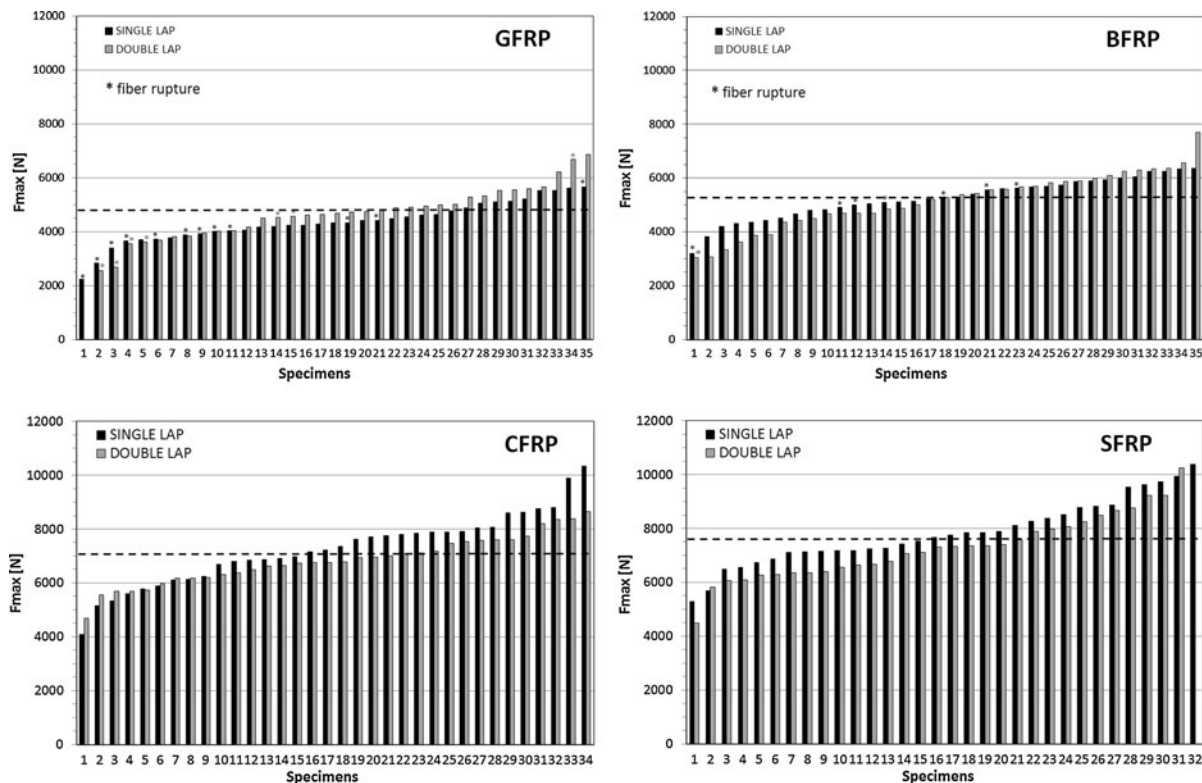
Figure 29 shows all results in terms of maximum load ( $F_{max}$ ) for each type of reinforcing material, grouping the data obtained from SL and DL tests in ascending order and discarding anomalous values obtained for steel reinforcements (Table 8). The average values, computed without taking into account the results of the specimens for which fiber rupture occurred, are also reported.

Figure 30 shows all results obtained for the four reinforcing materials (both SL and DL tests). Failure modes of specimens (debonding or fiber rupture) are distinguished.

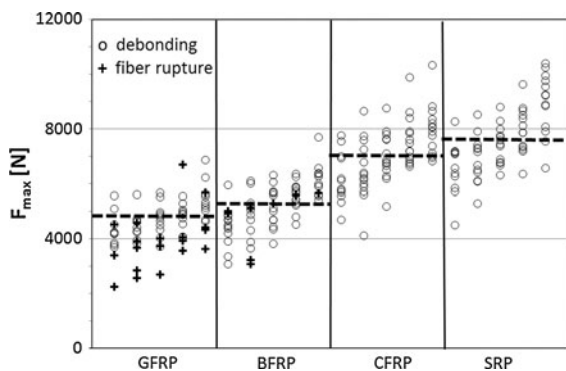
Lastly, Fig. 31 compares the envelopes of the representative load–displacement curves. Specimens reinforced with GFRP, BFRP and CFRP show similar global behavior, particularly in terms of maximum displacements; the GFRP and BFRP reinforcing systems also show good agreement in terms of maximum load. Specimens reinforced with SRP are characterized by the greatest strength, but the maximum displacement is quite low, due of lack of the horizontal part. In this case, the load cannot be transferred properly, as the effective transfer length is probably higher than the bonding length used in the tests.

This data processing allows us to make the following comments:

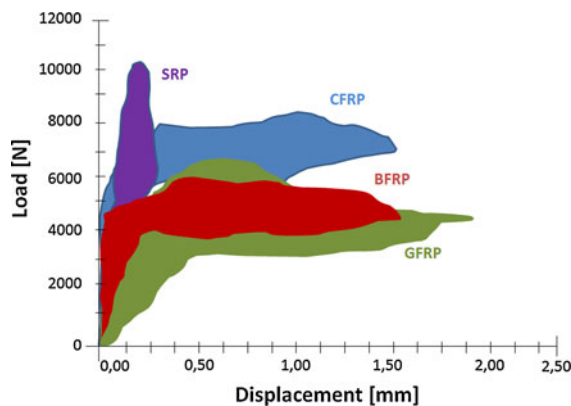
- Fiber rupture only affects specimens strengthened with the GFRP and BFRP systems. In particular, most of the cases of fiber rupture were those strengthened with GFRP (19 specimens, 27 % of the total of 70 tests). This behavior is due to the fact that the average debonding load of GFRP is about 60 % of the tensile failure load, so that, when stress concentration occurs due to experimental imperfections, tensile failure takes place instead of debonding.
- Comparing the average value of the failure load of each type of reinforcing material, the specimens reinforced with GFRP strips had the lowest values (average 4,747 N), while those strengthened with SRP had the highest ones (7,568 N). Nevertheless, specimens strengthened with BFRP had an average maximum load (5,261 N) similar to that of specimens strengthened with GFRP, and those strengthened with CFRP had an average maximum load (7,048 N) similar to that of specimens strengthened with SRP. It is worth noting that higher axial stiffness of the reinforcement gives a higher debonding load, due to an increasing of the effective transfer length.
- The results in terms of debonding load ( $F_{max}$ ) deduced from the SL and DL tests are in accord for all types of EB systems within acceptable limits, although the values deduced from SL are often larger than those of DL. This may depend on the non-equal distribution of the load between the two sides of DL specimens, which occurs mostly for stiffer materials (CFRP and SRP), perhaps due to defects in the accommodation of the reinforcement



**Fig. 29** Overall results in terms of maximum load for each reinforcing material. *Dotted lines* average values deduced by discarding specimens in which fiber rupture occurred (marked by asterisk)



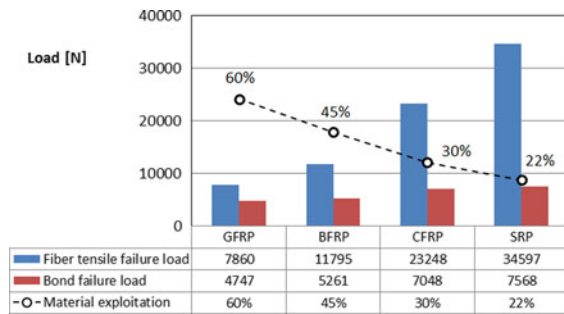
**Fig. 30** Comparison among all results in terms of maximum load (*dotted lines* average values deduced by discarding specimens in which fiber rupture occurred)



**Fig. 31** Load–displacement curves envelope for all reinforcing materials

round the roller device. Conversely, for GFRP strengthening, SL showed lower values than DL, probably due to the greater difficulty in providing perfect alignment of the load in the SL device, which may induce stress concentration in the bond.

- Computing the ratio between the average value of the debonding load and the tensile failure load of the strips (Table 4), a measure of the exploitation of material strength was derived (Fig. 32): 60 % for glass, 45 % for basalt; 30 % for carbon, and 22 % for steel reinforcements.



**Fig. 32** Comparison between fiber tensile failure load and debonding load, with exploitation of material strength

This confirms the fact that, because GFRP and BFRP specimens have lower strength in comparison with CFRP and SRP ones, their mechanical properties are more exploitable by the substrate than the high-performance materials. However, greater strength reinforcements increase the debonding load, thanks to their higher stiffness.

#### 4 Conclusions

Investigation of the bond characteristics of four composite EB systems (GFRP, BFRP, CFRP, SRP) applied to the surface of masonry bricks was performed by means of a RRT involving 12 laboratories from European universities and research centers. Monotonic tests were carried out, with different set-ups for SLST and DLST. Out of the total of 280 specimens, half were tested in SL, half in DL (with two different set-ups, DL55 and DL110). The results gave much information on several aspects, as follows:

Failure mode and specimen preparation:

- Debonding failure mode was evidenced in most tests (except for GFRP and BFRP specimens, in which some problems of fiber rupture occurred), a thin layer of brick being detached together with the strip.
- The brick had a smoother front in comparison with the back; however, this difference in the surface texture was not reflected in peak load values.
- Strips should be impregnated carefully, especially in GFRP and BFRP applications, to ensure redistribution of stress among the fibers and to reduce anomalies in gauges performance.

Nevertheless, GFRP and BFRP may show fiber failure even when impregnated, due to their lower tensile strength, which is comparable to their debonding strength.

- The procedure established for preparing specimens was suitable for use even by different operators. No differences in the results were in fact detected among the specimens prepared at single laboratories (TECN, UMINHO and UPATRAS) and those provided to all the others.

Mechanical performance and effective transfer length:

- Specimens reinforced with GFRP and BFRP reached similar average maximum loads at debonding (4,747 and 5,261 N, respectively), lower than the values for specimens reinforced with CFRP and SRP (7,048 and 7,568 N, respectively).
- The CoV of debonding loads was similar among the four reinforcing materials (total average values ranging of 14–17 %). This dispersion may be compared with the tensile strength of the single reinforcing materials (CoV range 7–13 %) and stiffness (CoV range 5–10 %), or the brick pull-off strength (CoV about 12 %).
- Activation of the effective transfer length was confirmed by almost flat overall behavior, once the maximum load had been attained, shown in the load–displacement diagrams. This was detected for specimens reinforced with GFRP, BFRP and CFRP; only specimens reinforced with SRP showed overall elastic brittle behavior, thus confirming that the effective transfer length was higher than the adopted bonded length (160 mm).
- In particular, the effective transfer length was lower for specimens reinforced with GFRP and BFRP (about 80 mm) in comparison with those reinforced with CFRP (between 120 and 160 mm) and with SRP (higher than 160 mm), due to the lower stiffness of the fibers.
- Exploitation of fiber strength (i.e., ratio between tensile failure load and debonding load) was higher for specimens reinforced with GFRP and BFRP (45–60 %) than those reinforced with CFRP and SRP (22–30 %); in GFRP and BFRP specimens, the debonding load was closer to the tensile strength of the reinforcement.

### Measurement system and test set-up:

- The lay-out of the strain gauges allowed deriving the strain profiles and the  $\tau$ -slip behavior. Extra displacement transducers were useful in providing global load–displacement curves.
- However, in many cases, the data recorded by the strain gauges and displacement transducers were incorrect or difficult to interpret, due to various problems such as misalignment of fibers at the measuring points or defects in the adhesion of the transducer supports to the strips.
- In the case of DLST, the stress evaluation provided by the gauges on both sides of the bricks displayed differences, apparently related mainly to local effects, such as the non-uniform distribution of stresses in the fibers, rather than to the unequal distribution of load between the two sides of the brick.
- In some cases, comparisons of the slip obtained by integrating the strain profile along the strip also showed some non-negligible differences, which may have depended on brick strain, micro-displacements during loading neglected in integration, or the above-mentioned non-uniform strain distribution in the fibers.
- The DL set-up, especially designed and agreed upon among partners, turned out to be easily implemented in the laboratory with common electromechanical or servo-hydraulic universal machines; it allowed simple self-equilibrated tests to be carried out, and the load can be assumed to be evenly distributed between the two composite strips, due to mechanical constraints (e.g., ball joints and roller device).
- DL55 and DL110 set-ups showed very similar results: considering each single reinforcing material, the variation of mean loads ranged from 0.2 to 9 % between the two systems; except for CFRP, DL110 showed results slightly more scattered than DL55.
- Therefore, the higher curvature of the strip adopted in DL55 set-up did not influence significantly the results. Proper impregnation of the curved unbonded length, also helps to avoid stress concentration.
- Testing efficiency may be affected by different boundary conditions, ought to modifications made on the DL test set-up. Nevertheless, in the

present experiments, only slight effects were observed on the results, in case of lacking of the hinge at the base. Of course, care in centering the specimen correctly is particularly recommended.

- SL set-ups require specific testing frames and care in load alignment, but are not affected by uncertainties in the values of load applied to the strip. Results showed higher sensitivity to the brittleness of fibers with respect to DLST.
- Nevertheless, in the ambit of each single reinforcing material, mean values of debonding loads differ in a small range (2–7.5 %) between SL and DL set-ups.
- The SL system also requires particular care in clamping the free end of the strip. Although several systems were adopted by the various laboratories, no particular differences in results were recorded.

The presented experiments investigated the bond efficiency of EB composites applied to masonry bricks, with the aim of clarifying the influence of various reinforcing systems and test set-ups. A good match between the two set-ups and good repeatability in various laboratory conditions were found. Although the composite-to-brick bond is probably the major contributor to the behavior of external bond strips applied to brick masonry, it is worth recalling that other factors can also play an important role in evaluations, such as: the presence of mortar joints, misalignment of bricks, substrate and environmental conditions, and care in application. In this perspective, further experiments are needed, to investigate higher bond lengths (especially for CFRP or SRP), various composite material thicknesses; various substrates (e.g., stone) with differing mechanical properties; differing properties of adhesives; the presence of mortar joints (by tests on masonry prisms); cyclic behavior; and the influence of hydro-thermal and aggressive environmental conditions (e.g. salty). Some of this work has already been planned within the framework of TC 223-MS as an extension and/or integration to the understanding of the bonding phenomenon.

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